# Investigation of Correlations for Prediction of Critical Properties of Pure Components

# A Thesis

# Submitted to the College of Engineering of Nahrain University in Partial Fulfillment of the Requirements for the Degree of Master of Science in Chemical Engineering

# by

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#### ABSTRACT

Prediction of accurate values of critical properties of any pure component is very important because they are often utilized in estimating the physical properties for chemical process design. Experimental measurements of critical properties for components are very difficult. So, in order to obtain accurate critical property values, attention has been turned to calculate it using mostly a group contribution method which is difficult. To overcome this problem efforts where directed to modify or improve equations to calculate critical properties using relatively simple method.

In this method, critical temperature, critical volume and critical pressure can be estimated solely from data of the normal boiling point and molecular weight of pure substance by means of successive approximations that are repeated until calculation of critical pressure ( $P_C$ ) converges.

The procedure can be summarized as follows:

- 1. Calculation of critical temperature.
- 2. Assuming suitable value of critical pressure and calculating critical volume.
- 3. Calculation of critical compressibility factor.
- 4. Calculation of critical pressure from the following equation ;

$$P_c = \frac{Z_c R T_c}{V_c}$$

If  $P_{\rm C}$  obtained is different than  $P_{\rm C}$  assumed, the procedure repeated until  $P_{\rm C}$  obtained is the same as  $P_{\rm C}$  assumed with tolerance of 10<sup>-4</sup>.

 A statistical program was used to obtained the following equation to estimate critical temperature for non – polar and polar compounds using the normal boiling point and molecular weight of pure substance ;

$$Tc = -11.5565 - 1.03586Mwt + 2.075167Tb - 0.000281Mwt^{2}$$
$$-0.00131Tb^{2} + 0.001827MwtTb$$

The AAD% for estimation of critical temperature is 0.9878 % for 114 nonpolar compounds and 1.3525 % for 16 polar compounds.

2. Reidel equation for estimation of critical volume was used after assumption of suitable value for critical pressure. The AAD% of estimation critical volume in four steps method is 1.7651 % for 78 compounds of various chemical groups.

3. Pitzer equation for critical compressibility factor was used for non-polar compounds and the following modified equation of Wu and Steil was used for the critical compressibility of polar compounds;

$$Z_c = 0.2901 - 0.879 * \omega - 0.033583 Y$$

Lee – Keseler method for calculating the acentric factor was used.

4. The results of the values of critical pressures prediction by this four steps procedure proved to be very satisfactory. The AAD% for 114 non-polar compounds is 1.98% and the AAD% for 16 polar compounds is 1.4671%.

5. The above results indicate that the proposed four steps method is more accurate than other methods including group contribution methods; this is in addition to the fact that it is relatively easy method when it is compared to the complicated group contribution method.

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## Nomenclature

Α	Molecule surface area
$\mathbf{A}_{+}$	Amount of positively charged surface area
A_	Amount of negatively charged surface area
A <sub>HB</sub>	Amount of charged
A,B,C	<b>Constants in Nokay correlation</b>
a	<b>Constant in Forman-Thodos correlation</b>
	$(cm^{6}.atm/(gmole)^{2})$
b	<b>Constant in Forman-Thodos</b>
	correlation(cm <sup>3</sup> /gmole)
<b>c</b> <sub>1</sub>	Constant in correlation of Mathur et.al.
f	Holmholtz free-energy
Mwt	Molecular weight
N <sub>atoms</sub>	Number of atoms
Р	Pressure (bar)
P <sub>C</sub>	Critical pressure (bar)
Pr	Reduced pressure
Рск	Group contribution for critical pressure by
	Joback
P <sub>VP</sub>	Vapor pressure (bar)
R	Universal gas constant (cm <sup>3</sup> .bar/mole.K)
S	Entropy (J/mole.K)
Sp.Gr	Specific gravity at 60 °F (15.56)
Т	Temperature (K)
T <sub>b</sub>	Normal boiling point (K)
T <sub>C</sub>	Critical temperature (K)

T <sub>r</sub>	Reduced temperature
t <sup>*</sup>	At standard condition
$\Delta_{\mathrm{T}}$	Group contribution for critical temperature
U	Internal energy (J/mole)
V	Volume (cm <sup>3</sup> /mole)
V <sub>C</sub>	Critical volume (cm <sup>3</sup> /mole)
V <sub>r</sub>	Reduced volume
V <sub>lb</sub>	Volume of liquid at boiling point (cm <sup>3</sup> /gmole)
Δv	Group contribution for critical volume
V <sub>CK</sub>	Group contribution for critical volume by Jobock
ω	Acentric factor
Y	Polarity factor
Z <sub>C</sub>	Critical compressibility factor
Z	<b>Compressibility factor</b>
ρι	Density of liquid (g/ cm <sup>3</sup> )
α	Riedel factor
θ <sub>r</sub>	$T_b/T_C$

# **CHAPTER ONE**

#### Introduction

Thermodynamic properties correlation developed in the past 50 years and based on the principle of corresponding states.

Critical properties are related to chemical process design. Having said that the vapor-liquid critical point is at best weakly defined in comparison with say the triple point. In fact it is best to think of it not a point but a region of undefined boundaries where the stability of the system breaks down and the response to a small turbulence is massive in comparison to the size of the disturbance. The practical need is for knowledge of the critical properties of as much substance as possible. Then, corresponding states correlations can be devised and applied to estimate properties away from critical point. However, the number of substance that is stable at their critical point is limited.

Critical properties are essential in estimating thermodynamic and volumetric properties by the theorem of corresponding states. They are also of importance in determining the existing phase conditions in or the permissible operating ranges of both reactors and mass transfer equipment. As critical properties are often difficult to measure experimentally, reliable method for correlating and predicting them are necessary.

The principle aim of this work is to study the existing methods for calculating the critical properties (critical volume, critical temperature, critical pressure and critical compessibility factor), and to present the results of these methods for critical properties that will show which of these methods are more suitable and rliable than others. Efforts will also be directed to modify or improve the best method if possible in order to come out with a method, or methods, that may predict the critical properties for pure components with high accuracy.

## **CHAPTER TWO**

#### **Theories and Definitions**

#### 2.1 Thermodynamic of the Critical Point

:

The critical state of a fluid is represented by the point on P, V, T surface where the volume of the gas and liquid phases become identical.

Such a point may be said to lie on the border separating the stable and unstable parts of a continuous P, V, T surface[8].

A discussion of the behavior of thermodynamic functions at and near this point may, therefore, be based on either of the two surfaces U,S,V or F,V,T.

Liquid and gas can exist together at equilibrium if the temperature, pressures and chemical potential of the two phases are equal. These qualities may be written in terms of the molar Helmhohz Free – energy of a one component system, as follows

 $T^{l} = T^{g} \tag{2.1}$ 

$$\left(\frac{\partial f}{\partial \upsilon}\right)_{T}^{l} = \left(\frac{\partial f}{\partial \upsilon}\right)_{T}^{g}$$
(2.2)

$$f^{l} - \upsilon^{l} \left(\frac{\partial f}{\partial \upsilon}\right)_{T}^{l} = f^{g} - \upsilon^{g} \left(\frac{\partial f}{\partial \upsilon}\right)_{T}^{g}$$
(2.3)

Thus, a graph of f as a function of v at constant temperature is represented by the full – line in Fig 2.1.



Fig. 2.1 The Holmholtz free-energy as a function of volume[33]

The curve on the P,V,T surface which corresponds to the dashed curve in Fig 2.1 is shown in Fig 2.2.



Fig. 2.2 the pressure as a function of volume[33]

The equality of the chemical potential of the two phases requires that [26],

$$\int_{ABCD} \partial P = 0 \tag{2.4}$$

Thus, the straight line AD cuts the continuous curve to form two equal areas above and below the transition pressure.

A critical point occurs on a free – energy surface if the tangent AD becomes vanishingly small, so that the four points A,B,C and D coincide . At such a point,

$$\left(\frac{\partial f}{\partial \upsilon}\right)_T < 0$$
 ,  $\left(\frac{\partial^2 f}{\partial \upsilon^2}\right) = 0$  ,  $\left(\frac{\partial^3 f}{\partial \upsilon^3}\right) = 0$  (2.5)

That is

$$P > 0 , \left(\frac{\partial f}{\partial v}\right)_T = 0 , \left(\frac{\partial^2 p}{\partial v^2}\right) = 0 , \left(\frac{\partial^3 p}{\partial v^3}\right) < 0$$
(2.6)

The second derivative and third derivative expressed in eq. (2.5) are necessary if B and C are to coincide, and the fourth is the condition that the fluid should be stable at volumes immediately above and below the critical point. The fluid is stable at this point as the pressure falls with increasing volume, but the stability is of lower order than that of other states of the fluid.



Fig. 2.3 The P-V behavior of the fluid[33]



Fig. 2.4 The P-T behavior of the fluid[33]

#### 2.2 Critical Phenomena in Liquids

If a liquid, such as water, is sealed in an evacuated tube, a certain amount will evaporate to form vapor. This vapor will exert a pressure just as any gas does, and provided the temperature is maintained constant, and equilibrium will be established between the liquid and vapor phases. The vapor pressure established is characteristic for each liquid and is constant at any given temperature; it is known as the saturated vapor pressure of the liquid. The saturated vapor pressure increases continuously with temperature. Thus, at 25 °C the vapor pressure of water is 23.76 mmHg, while at 100 °C it is 760 mmHg. As the water in the sealed tube is heated further, more and more water evaporates and the pressure continues to increase. At all times there is a definite line of demarcation, or meniscus, between the liquid and vapor phases. When the temperature of 374 °C is reached, however, the meniscus becomes indefinite, fades into the vapor, and disappears. At this temperature the physical properties of liquid and vapor become identical, and no designated can be observed between the two. A liquid in this condition is said to be at the critical point. The temperature, saturated vapor pressure, and molar volume corresponding to this point are designated the critical temperature, critical pressure, and critical volume respectively. Their values, which are constant and characteristic for each substance, are known as the critical constants. On heating the sealed tube even slightly above the critical temperature, no evidence can be found of the presence of liquid. The whole mass is gaseous and remains in that state no matter how high it is heated, or how large an external pressure is applied. Since the phenomena described for water is exhibited by all liquids, it must be concluded that no liquid can exist at such temperature above the critical under any applied pressure.

The critical phenomena is reversible, when the gas in the tube is cooled below the critical temperature, if the pressure is sufficiently high, the meniscus reappears, and again we have the two phases, liquid and vapor[26].

#### **2.3 The Boiling Point of Liquids**

The normal boiling point of a liquid is the temperature at which the vapor pressure of the liquid equal 760 mm Hg pressure, or 1 atm. However, a liquid can be made to boil at any temperature between its freezing point and the critical temperature by merely raising or lowering, as the case may be, the external pressure on the liquid. Therefore, it may be stated in general that boiling point of a liquid is the temperature at which the vapor pressure of a liquid becomes equal to the external pressure acting upon the surface of the liquid. Boiling is characterized by the formation within the liquid of bubbles of vapor, which rise and escape into the vapor phase[28].

#### **2.4 Critical Temperature**

The critical temperature is that temperature above which unique liquid and gas phases do not exist. As approach the critical temperature, the properties of the gas and liquid phases become the same, so above the critical temperature there is only one phase. The critical pressure refers to the vapor pressure at the critical temperature. The critical molar volume is the volume of one mole of material at the critical temperature and pressure[30].

#### **2.5 Three Parameter Correlations (non-spherical molecules)**

In general, successful equations of state have included one or more dimensionless characteristic parameters in to the function.

The first step in accomplishing this is to introduce a third parameter; usually it is related to the vapor pressure, or volumetric property at or near the critical point. Historically, several different third parameters were introduced at about the same time but the most popular have been  $Z_C$  (Lyderson 1955) and the acentic factor  $\omega$  (Pitzer and Curl 1955)[18].

Lyderson and later revision by Hougan, et al., (1959) tabulated Z (and reduced thermodynamic properties) at increment of  $T_r$  and  $P_r$  for different values of  $Z_C$  in practice, this correlation has been used only occasionally, such as by Edward and Thodos (1974) for estimating saturated vapor densities of non polar compounds[28].

#### **2.5.1**Critical Compressibility Factor

The critical compressibility factor was first proposed to be used as a third parameter by Lydersen, Hougen, and GreenKorn, and defined as:

$$Z_c = \frac{P_c V}{RT_c} \tag{2.7}$$

The value of Zc varies from about 1.23 for water to 0.3 for low hydrocarbons[33].

The compressibility factor of a polar fluid can expressed as a linear relationship in the  $\omega$  and polarity factor Y as following equation[36],

$$Z_c = 0.2901 - 0.0879 \,\omega - 0.00266 \,\mathrm{Y} \tag{2.8}$$

Most polar compounds have polarity factor in the range of 0.24 to 0.26, while hydrocarbons are normally in the 0.26 to 0.28 range.

Table 2.1 shows the values of w and Y for polar fluids.

Substance	ω	Y
Ammonia	0.252	0.739
Acetone	0.305	0.652
Ethylene Oxide	0.202	0.565
Hydrogen Chloride	0.126	0.565
Methyl Floride	0.191	0.739
Methyle Chloride	0.152	0.304
CH <sub>2</sub> F <sub>2</sub> (R-32)	0.278	0.696
Water	0.344	1.00
Ethanol	0.641	0.213
Methanol	0.559	0.652
Iso-Propanol	0.663	0.0521
n-Propanol	0.626	-0.0521

Table 2.1 Values of ω and Y for polar fluids

#### 2.5.2 A centric Factor

Pitzer in 1955[18] postulated that the slope of the reduced vapor pressure against reciprocal of reduced temperature curve is the most sensitive property for a third parameter base. Since vapor pressure can be measured with grater accuracy than critical properties, this approach should be superior to the critical compressibility factor. Thus he defined his acentric factor  $\omega$ , as:

$$\omega = -\log(P_r^{sat})_{Tr=0.7} - 1$$
 (2.9)

Where  $p_r^{sat}$  is the reduced saturated vapor pressure at reduced temperature  $(T_r=0.7)$  (i.e., a temperature near the normal boiling point). This form is chosen to make  $\omega = 0$  for simple fluids like (Ar, Kr, and Xe) with simple spherical molecules. Hence acentric factor is a factor for measuring the acentricity or non sphericity of a molecule. However, it should be noted that  $T_r = 0.7$  is close to the normal boiling point of most substances, thus the particular choice of  $T_r = 0.7$  adopted by Pitzer provides not only numerical simplicity because log  $p_r^{sat} = 1.0$  but also convenience because vapor– pressure data are most commonly available at pressure near atmospheric.

This latter estimation would normally be made by using one of the reduced vapor pressure correlations, for example

$$\log P_{vp} = A + \frac{B}{T} \tag{2.10}$$

With A and B found , from the sets  $(T_C, P_C, T_b, P=1)$ , then

$$\omega = \frac{3}{7} * \frac{\theta}{1 - \theta} \log P_C - 1 \tag{2.11}$$

Similarly, if the Lee- Kesler vapor pressure relations are used

$$\omega = \frac{\alpha}{\beta} \tag{2.12}$$

Where

$$\alpha = -\ln P_c - 5.97214 + 6.09648 \,\theta^{-1} + 1.28862 \,\ln \theta - 0.169347 \,\theta^6 \quad (2.13)$$

$$\beta = 15.2518 - 15.6875\theta^{-1} - 13.472\ln\theta + 0.43577\theta^{6}$$
(2.14)

#### **2.6 Reduced Properties**

The most generalized correlations are based on the observation that data for different fluids exhibited a remarkable uniformity when the thermodynamic coordinates are expressed in suitable dimensionless or reduced form. This fact is experimental basis of the theorem of corresponding states[40]. Thus the reduced conditions are

$$T_r = \frac{T}{T_C} \tag{2.15}$$

$$P_r = \frac{P}{P_c} \tag{2.16}$$

$$V_r = \frac{V}{V_C} \tag{2.17}$$

#### 2.7 Universal gas constant

As the pressure approaches zero, the molecules was separated by the infinite distances. Their volumes become negligible compared with the total volume of the gas, and the intermolecular forces approach to zero. At the conditions all gases was said to be ideal, the proportionality constant is called universal gas constant. Its numerical value is determined by means of the following equation:

$$R = \frac{(P \ V)_{t^*}}{273 \ .15 \ K} \tag{2.18}$$

Since PVT data are not in fact be taken at zero pressure, data are taken at infinite pressure are extrapolated to the zero pressure state[40]. The accepted value of  $(PV)t^*$  is 22,7118 cm<sup>3</sup>.bar.mole<sup>-1</sup>, leading to the following value of R:

$$R = \frac{22,711.8 \,\mathrm{cm}^3.\mathrm{bar.mole}^{-1}}{273.15K} \tag{2.19}$$

#### **CHAPTER THREE**

#### **Literature Survey**

Correlation attempts on the critical properties of pure liquids are numerous and have been reviewed by several authors

#### **3.1 Reidel Method**

Reidel [3] proposed the following correlation for  $V_c$ ;

$$V_c = \frac{RT_c}{P_c} [3.72 + 0.26(\alpha - 0.7)]^{-1}$$
(3.1)

where

$$\alpha = 0.9076 \left[ 1.0 + \frac{(T_b / T_c) \ln P_c'}{1.0 - (T_b / T_c)} \right]$$
(3.2)

$$P_c' = P_c / 1.01325(10^5)$$
 (3.3)

#### 3.2 Method of Lyderesn

Lydersen [23]described a group-contribution method for estimating critical temperature, critical pressure, and critical volume utilizing atomic and structural increments for each multivalent atom or group in the molecule, his final equations are:

$$T_b / T_c = 0.567 + \sum \Delta_T - (\sum \Delta_T)^2$$
(3.4)

$$P_c = 0.101325M / (\sum \Delta_{\rm P} + 0.34)^2$$
(3.5)

$$V_c = (2.497 + 0.06243 \sum \Delta_v) / M \tag{3.6}$$

#### **3.3 Forman and Thodos**

Forman and Thodos [9] devised a group contribution method which depends on structure and estimated suitable Van der Waals' constants (a and b) which were correlated with molecular structure.

Then their equations are;

$$T_{c} = 8 a / 27 bR \tag{3.7}$$

$$P_c = 0.101325 * a/27b^2 \tag{3.8}$$

#### **3.4 Nokay Method**

Nokay [27] had correlated critical temperature by

$$\log T_c = A + B \log(Sp.Gr.) - C \log T_b$$
<sup>(3.9)</sup>

Table 3.1 Summery of revised constants for use in Nokay's equation forpredicting critical temperature

Family of compounds	А	В	С
Alkanes ( Paraffine )	1.359397	0.436843	0.562244
Cycloalkanes ( naphthenes )	0.568122	-0.071646	0.811961
Alkenes ( Olefins )	1.095340	0.277495	0.655628
Alkynes ( acetylenes )	0.746733	0.303809	0.799872
Alkadienes ( diolefins )	0.147578	-0.396178	0.994809
Aromatics	1.057019	0.227320	0.669286

#### 3.5 Viswanash Method

Viswanash [36] estimated  $V_c$  from  $T_c$  and  $P_c$  by the simple equation

$$V_c = (0.6243 + 0.259RT_c/P_c)/M \tag{3.10}$$

#### 3.6 Mathur et al. Method

Mathur et al., developed a simple relationship for critical temperature which for hydrocarbons is:

$$T_{c} = n_{1}(\log M) + C_{1}$$
(3.11)

Results of the evaluations are summarized in table 3.2. The method of Mathur et al. was the poorest of the correlations for estimating critical temperature, Lydersen and Forman-Thodos were almost equivalent in the first evaluation.

	Mathur et al.		Forman-Thodos		Lydersen	
Chem.famialy	Avg.dev.(K)	Data point	Avg.dev.(K)	Data point	Avg.dev.(K)	Data point
Paraffins	4.77	20	7.55	78	3.12	86
Naphthenes	11.0	27	8.24	35	8.32	36
Olefins	4.73	9	3.74	34	6.62	34
Diolefins			12.08	9	8.20	9
Acetylenes	18.07	7	9.18	14	8.10	14
Aromatics	5.56	5	7.01	24	5.20	26
Overall Avg.dev	8.66	68	7.27	194	5.44	205

 Table 3.2 Comparison of the methods for prediction of critical temperature

Results of the evaluations of critical pressure are summarized in table 3.3.for saturated compounds and aromatics, the methods are essentially equivalent with Lydersen. However Forman-Thodos method is appreciably better for all classes of unsaturated compounds.

Lydersen			Forman-Thodos		
Avg.dev.(MN/m <sup>2</sup> )		Data point	Avg.dev.(MN/m <sup>2</sup> )	Data point	
Chem. Family	y	_	_	_	
Paraffins	0.0949	86	0.0945	78	
Naphthenes	0.1089	35	0.1401	35	
Olefins	0.1159	34	0.0866	34	
Diolefins	0.3233	9	0.0931	9	
Acetylenes	0.2044	14	0.1435	14	
Aromatics	0.1176	24	0.1326	24	
Overall Avg.dev	0.1213	202	0.1096	194	

Table 3.3 Comparison of the methods for prediction critical pressure

Results of the evaluations of critical volume are summarized in table 3.4. With the correlation of Reidel showing the greatest accuracy for saturated and aromatics with the correlation of Lydersen having a slight advantage for unsaturated hydrocarbons.

Table 3.4 Results of the evaluating of methods for predicting the criticalvolume of pure hydrocarbon

Avg. Deviation, m <sup>3</sup> /Kg. * 10 <sup>3</sup>						
Ν	No. of data	Lydersen	Viswanath	Riedel	Hall-	
points					Yarborough	
Chem. family						
Paraffins	78	0.1205	0.1617	0.0868	0.1636	
Naphthenes	6	0.0306	0.0606	0.0266	0.0643	
Olefins	13	0.1673	0.1654	0.1785	0.2260	
Diolefins	2	0.0537	0.0387	0.0581	0.1779	
Acetylenes	5	0.0325	0.0755	0.0624	0.2628	
Aromatics	22	0.0925	0.0874	0.0661	0.1105	
Overall	126	0.1120	0.1386	0.0888	0.1607	
Avg.dev						

#### 3.7 Leonard I.Stiel and George Thodos Method

Frost and Kalkwarf[5] developed the following equation for the vapor pressure of pure substance which is applicable between the triple and critical point:

$$\log p = A + \frac{R}{T} + C\log T + D\frac{p}{T^2}$$
 (3.12)

Where A,B,C and D are constants specific to the individual substances.

Reynes and Thodos had expressed equation (3.12) in the following reduced form:

$$\log p_{R} = a + \frac{\beta}{T_{R}} + \gamma \log T_{R} + 0.1832 \frac{p_{R}}{T_{R}^{2}}$$
(3.13)

They applied Eq.(3.13) to 54 hydrocarbons of all types and they found the following relationship to exist between the constants  $\beta$  and  $\gamma$  for these substances:

$$\beta = \frac{5}{9}(\gamma - \frac{8}{3}) \tag{3.14}$$

These constants can be determined for any substances from its vapor pressure data using the procedure suggested by Frost and Kalkwarf. They showed that the constant C is the slope, and the constant B, is the intercept of the equation (3.15)

$$Y = CX + B \tag{3.15}$$

The resulting relationship is presented in figure 3.1 and can be expressed analytically as follows:

$$T_c = \frac{B}{0.5643^* C - 1.452} \tag{3.16}$$

Thus, equation (3.5) permits the establishment of the critical temperature of any pure substance from its available vapor pressure data.

At critical point, the Frost – Kalkwarf equation becomes,

$$\log P_c = A + \frac{B}{T_c} + C\log T_c + D\frac{P_c}{T_c^2}$$
(3.17)

from Van der Waal's constant

$$D = \frac{a}{2.303 R^2}$$
(3.18)

where R is the universal gas constant. Therefore, the following expression results for critical pressure:

$$\log P_c = A + \frac{B}{T_c} + C \log T_c + 0.1832$$
(3.19)

critical temperatures and pressures for 64 non-hydrogen bonding substance including saturated and unsaturated aliphatics, naphthenes, aromatics, alkyl halids, monatomic and diatomic gases, sulphur dioxide, and nitrous oxide were calculated by the approach outlined above using experimental vapor pressure data obtained from literatures. The resulting average deviations were 0.59% for critical temperature and 3.18 % for critical pressure.

#### **3.8** New Simple Correlation Pridicts Critical Temperature

They showed how several transport and thermodynamic properties of organic liquids can be correlated with the molecular weight to within an average error of 5% from experimental values. They also presented a similar method for predicting the critical temperature within an overall average error of 0.78%. They used two equations for hydrocarbons and related compounds [25],

$$T_c = n_1(\log M) + C_1$$
 (3.20)

For other organic liquids,

$$T_c = C_2(M)^{n^2}$$
 (3.21)

For hydrocarbons and related compounds, the linear relationship was obtained in semilogarithmic coordinates; for other organics, in logarithmic ones. Data for 219 organic liquids –with molecular weights between 30 and 590 and with critical temperatures between 190 and 905 °K –were used.

The values of the constants for both equations are given in table (3.5), as well as average percent error and the number of compounds considered in their study. Table (3.6) shows only a small section of the much larger complete tabulation.

Mathur-Ibrahim-Kuloor method								
series	No.Compounds	Value of	constants	Avg. %				
	Considered	<b>n</b> <sub>1</sub>	<b>c</b> <sub>1</sub>	error				
hydrocarbo	hydrocarbons and related compounds,Eq.(1)							
paraffins	20	497.5	-453.0	0.68				
noco-oiefins	19	490.0	-441.0	0.78				
alkynes	10	465.4	-370.8	0.82				
aikyibenzenes								
up to C <sub>25</sub>	20	454.5	-309	0.67				
aboveC <sub>25</sub>	14	227.4	+255.7	0.33				
Cyclohexanes								
up to C <sub>25</sub>	20	500	-428	0.47				
aboveC <sub>25</sub>	17	304.1	+63.9	0.27				
Cyclopentanes								
up to C <sub>25</sub>	21	454	-333	0.54				
aboveC <sub>25</sub>	13	278.4	+112.7	0.24				
Other organic liquids , Eq. (2)								
		<b>n</b> <sub>2</sub>	<b>C</b> <sub>2</sub>					
alcohols	8	0.2525	191.4	1.22				
isoalcohols	4	0.2820	161	0.76				
ketones	4	0.2251	205.5	0.59				
Acids	4	0.1876	274.4	0.58				
Esters	19	0.3259	123	1.06				

Table 3.5 Values of constants  $n_1$ ,  $C_1$  for Eq.(3.20), and  $n_2$ ,  $C_2$  for Eq.(3.21)

Semesters	15	0.3048	133.7	0.91
Ethers	11	0.5469	46.3	2.55
total No =	219	overall avg . = 0.78		

# Table 3.6 Comparison of experimental and calculated values of critical

temperatures by Mathur method

		Tc,	Tc, °K	
Compound	Mwt	Experimental	calculated	Avg. %
				error
Normal paraffins				
Methane	16.04	190.6	46.4	-75.6
Ethane	30.07	305.4	282.3	7.6
Propane	44.09	369.8	365.0	1.29
Butane	58.12	425.1	424.9	0.047
Pentane	72.15	469.4	471.4	0.420
Hexene	86.17	507.3	510.1	0.550
Normal Mono-Olefin	S			<u>.</u>
Ethene(ethylene)	28.05	282.9	268.5	-5.000
Propene(propylene)	42.07	364.9	354.7	-2.800
Butene	56.10	419.0	416.0	-0.710
Pentene	70.13	461.0	463.4	0.520
Hexene	84.15	501.0	502.2	0.230
Alkynes				
Ethyne (acetylene)	26.03	309.3	287.9	-6.910
Propyne (methyl Acetylene)	40.06	394.6	375.0	-4.96
n-Hexyne	82.14	521.0	520.2	-0.0153

n-Heptyne	96.16	553.0	522.0	-0.180

#### 3.9 Toshikatsu Hakuta and Mitsuho Hirata

Normal boiling points and liquid densities for very large numbers of compounds are given in the literature, and with rare exceptions these data are very accurate[12]. The proposed method used only these accurate data, and the critical temperature was calculated from normal boiling point by using the following equations proposed by Sato.

To estimate Tc, Sato divided substances in to five classes as follows:-

(A) Inorganic compounds [halogenated and sulfur compounds except elements and hydrogen compounds were classified in to  $B_1$  and  $B_2$ . carbon compounds were included in  $C_1$  and  $C_2$ ].

$$T_{c} [^{\circ}k] = 1075 T_{b} [^{\circ}C] + 473$$
 (3.22)

For B1: Halogenide and sulfur compounds.

$$T_c = 1.38 T_b + 450 - 11F$$
 (3.23)

Where F is the number of flucrine atoms per molecule.

**For B2:** Cylices, aromatics, naphthalenes and heterocyclic free of halogens and sulfur.

$$T_c = 1.38 T_b + 450 - r^{0.6} (0.3 T_b - 10)$$
 (3.24)

Where r is the ratio of noncyclic carbon atoms to total number of carbon atoms in the molecule.

For C1: Noncyclic carbon compounds of which Tb is below 30 °C.

$$Tc = 1045 T_b + 430 \tag{3.25}$$

**For C2:** Noncyclic carbon compounds of which Tb is equal to or greater than 30°C.

$$T_c = 1.08 \text{ Tb} + 446$$
 (3.26)

By using the critical temperature obtained above and a liquid density  $(\rho_L)$  at or below the normal boiling point  $(T_b)$ , molar volume of liquid  $(V_{lb})$  at normal boiling point  $(T_b)$  calculated from the following modified Goldhamer equation , which was also .

Used by Fishtine;

$$V_{lb} = \frac{M}{\rho_{l}^{\prime}} \left(\frac{T_{c} - T^{\prime}}{T_{c} - T_{b}}\right)^{n}$$
(3.27)

The values of n are listed in table (3.7) for different group of compounds. where values of exponent n are given in table (3.7), critical volume was estimated from modified Benson equation by using  $T_c$ ,  $V_{lb}$  and an assumed  $p_c$ ;

$$V_c = V_{Lb} (1.981 + 0.422 * \log P_c)$$
(3.28)

Acentric factor  $\omega$  was calculated from equation proposed by demister .

$$\omega = \frac{3}{7} \left( \frac{\log P_c}{(T_c/T_b) - 1} \right) - 1 \tag{3.29}$$

where  $T_c$  and  $P_c$  are the values used in previous steps. The correlation between critical compressibility factor  $Z_c$  and acentric factor w is given as the following equation ;
$$Z_c = 0.291 - 0.08 \,\omega \tag{3.30}$$

Eq.3.30) had been proposed by Pitzer.

Critical pressure was estimated by using the critical temperature, critical volume and critical compressibility factor in previous steps;

$$P_c = \frac{Z_c RT_c}{V_c}$$
(3.31)

Table (3.8) shows a summary of statistical data

Table (3.7) <b>J</b>	Recommended	l exponent f	for the (	Goldhammer	equation(	(eq.3.	.27)
----------------------	-------------	--------------	-----------	------------	-----------	--------	------

compound group	n
Alcohols and water	0.25
Hydrocarbons and ethers	0.29
All other organic compounds	0.31
All inorganic compounds except water	0.333

Average deviation%							
No.ofsubstances	T <sub>c</sub>	P <sub>c</sub>	Vc	Remarks			
240 for T <sub>c</sub> and P <sub>c</sub> 228 for V <sub>c</sub>	1.39	2.61	2.04	All substances			
153 for T <sub>c</sub> ,P <sub>c</sub> and V <sub>c</sub>	1.34	2.39	1.91	hydrocarbons			
68 for T <sub>c</sub> and P <sub>c</sub> ,	1.27	2.86	2.54	organic ompounds excepthydrocarbons			
19 for T <sub>c</sub> and P <sub>c</sub> , compounds14 for Vc	2.28	3.58	3.02	Inorganic			

Table (3.8) Summary of statistical data

# **3.10 Method of Joback**

Joback [17] reevaluated Lydersen's group contribution scheme, added several new functional groups, and determined new contribution, these relations for the critical properties are;

$$T_{c}(K) = T_{b} [0.584 + 0.965 \{ \sum_{k} N_{k}(tck) \} - \{ \sum_{k} N_{K}(pck) \}^{2} ]^{-1}$$
(3.32)

$$P_{c}(bar) = [0.113 + 0.0032 N_{atoms} - \sum_{k} N_{k}(pck)]^{-2}$$
(3.33)

$$V_c(cm^3 mot^3) = 17.5 + \sum_k N_k(vck)$$
 (3.34)

Where the contributions are indicated as tck, pck and vck. The group identities and Joback's values for contributions to the critical properties were given in tables. For  $T_b$ , a value of the normal boiling point, was needed, this may be taken from experiment or by estimation from the following equation;

$$T_b = M^{0.404} \sum_{k} N_k (tbbk) + 156.00$$
(3.35)

Where *M* is the molecular weight and  $N_k$  is the number of atoms of type *k* with contributions *tbbk*.

#### 3.11 Method of Constantionu & Gani (GC)

Constantinonu & Gani [28] developed an advanced group contribution method based on the UNIFAC groups, but they allowed for more sophisticated functions of the desired properties and also for contributions at a "second order" level. The functions give more flexibility to the correlation while the second order partially overcomes the limitation of UNIFAC which cannot distinguish special configurations such as isomers, multiple groups located close together, resonance structures, etc., at the "first order" the general (CG) formulation of a function f(F) of aproperty F is:

$$F = f \left[ \sum_{k} (F_{1i}) + W \sum_{j} M_{j} (F_{2j}) \right]$$
(3.36)

Where f can be alinear or non linear function,  $N_K$  is the number of first order groups of type k in the molecule;  $F_{1k}$  is the contribution for the first order group labeled 1k be the specified property, F;  $N_J$  is the number of second – order groups of type *j* in the molecule; and  $F_{2J}$  is the contribution for the second – order group label 2j to the specified property, F. The value of W is set bo zero for first - order calculations and Set to unity for second – order calculations for the critical properties, the CG formulations are:

$$T_{c}(k) = 181.128 \ln \left[ \sum_{k} N_{k}(tc1k) + W \sum_{j} M_{j}(tc2j) \right]$$
(3.37)

$$P_{c}(bar) = \left[\sum_{k} N_{k}(pc1k) + W \sum_{k} M_{j}(pc2j) + 0.10022\right]^{-2} + 1.3705$$
(3.38)

$$V_{c}(cm^{3}mol^{1}) = -0.00435 + \left[\sum_{k}(vc1k) + W\sum_{k}M_{j}(vc2j)\right]$$
(3.39)

# 3.12 Method of Marrero and Pardillo

Marrero –Marejon and Pardillo –Fontdevila[28] described a method for  $T_c$ ,  $P_c$  and  $V_c$  they called it a group interaction contribution technique or what is effectively a *bond contribution method*. They gave equations that use values from pairs of atoms alone, such as > C. and - N, or with hydrogen attached, such as  $CH_3$  – and  $-NH_2$ .

Their basic equations are:

$$T_{c} = T_{b} \left[ 0.5851 - 0.9286 \left( \sum_{k} N_{k} t c b k \right) + \left( \sum_{k} N_{k} t c b j \right)^{2} \right]$$
(3.40)

$$P_{c} = [0.1285 + 0.0059 N_{atoms} - \sum_{k} N_{k} pcbk]^{-2}$$
(3.41)

$$V_c = 25.1 + \sum_k N_k vcbk$$
 (3.42)

Where  $N_{atoms}$  is the number of atoms compound, N<sub>k</sub> is the number of atoms of type *k* contributions *tcbk*, *tpcbk* and *vcbk*. Marrero and Pardillo provided estimation method for  $T_{b_i}$ 

$$T_b = M^{-0.404} \sum_{k} N_k (tbbk) + 156.00$$
(3.43)

# 3.13 Method of Grigoras

An early molecular structural approach to physical properties of pure organic substance was proposed by Grigoras[28]. The concept was to relate several properities to the molecular surface areas and electrostatics as generated by combining quantum mechanical results with data to determine the proper form of the correlation. Grigoras related the critical properties to molecular properties via relations such as:

$$V_{c} = 2.217 A - 93.0$$
(3.44)  
$$T_{c} = 0.633A - 1.562 A + 0.427 A_{+} + 9.914 A_{HB} + 263.4$$
(3.45)

Where *A* is the molecular surface area,  $A_{-}$  and  $A_{+}$  are the a mount of negatively and positively charged surface area on the molecule and  $A_{HB}$  is the amount of charged. Surface area involved hydrogen bonding.

These relationships can be used to obtain other properties such as  $P_c$  by correlations such as:

$$P_{c} = 2.9 + 25.2 (T_{c} / v_{c})$$
(3.46)

# **CHAPTER FOUR**

# **Investigation and Development**

# 4.1 Investigation of the Equations

Critical properties can be calculated using many equations. Usually the accuracy of some equations to calculate the critical properties of pure components is not satisfactory.

It is well known fact that the evaluation of any correlation or prediction method is done by comparison of the results obtained from the correlation with those results obtained from the experimental data which will measure the accuracy of this prediction.

There are several equations of state that are used to describe the critical properties of pure components, such as Lydesen, Nokay, Thodos, Viswanath, Mathur Fetal and many other equations. These equations are the most important and accurate equations that are used to determine the critical properties of pure component.

# **4.1.1 Viswanath Equation**

Viswanath[36] proposed equation to estimate the critical volume for hydrocarbons by using the critical temperature ( $T_c$ ) and critical pressure ( $P_c$ ), which is simply :-

$$V_c = 0.6243 + 0.259 \frac{RT_c}{P_c}$$
(3.10)

The absolute average deviation obtained from above equation is shown in table 4.1

Chemical family	No. of Data point	AAD%
Paraffins	52	3.366
Olefins	14	5.0903
Dioelefins	2	5.2822
Acetylenes	2	4.5415
Aromatics	8	2.8025
Over all AAD%	78	3.8881%

Table 4.1 The AAD% of Viswanath from the experimental Data

# 4.1.2 Reidel Equation

Reidel [3] has proposed an equation for evaluating the critical volume (Vc) for hydrocarbons by the following equation:

$$V_c = \frac{RT_c}{P_c} [3.72 + 0.26(\alpha - 7.0)]^{-1}$$
(3.1)

Where

$$\alpha = 0.9076 \left[ 1.0 + \frac{\binom{T_b}{T_c} \ln P_c}{1.0 - \binom{T_b}{T_c}} \right]$$
(3.2)

The results obtained in using this equation to calculate the critical volume of hydrocarbons compound are not very satisfactory, however, they are better than when using Viswanath equation. The absolute average deviation obtained from above equation is shown in table 4.2

Chemical family	No. of Data point	AAD %
Paraffins	52	1.7647
Olefins	14	1.7638
Diolefins	2	0.995621
Acetylenes	2	1.8204305
Aromatics	8	1.3764
Over all AAD%	78	1.7651%

Table 4.2 The AAD% of Reidel from the experimental Data

# 4.1.3 Sato Method

Sato [12] divided substances into five classes: Inorganic compounds, Halogenide and Sulfur compounds, Cylices – Aromatics – Naphthalenes and Heterocyclic, Non cyclic carbon compound of which  $T_b$  is below 30 °C, Non cyclic carbon compounds which  $T_b$  is equal to or greater than 30°C. This method is not satisfactory because it is not one general method for all compounds, in addition to the fact that its accuracy is rather high .Some of the results obtained from this method are listed in table 4.3 for 2 classes only because the percentage absolute average deviation obtained from this method is very high as shown in table 4.3

Chemical family	No. of Data point	AAD %
Sulfur compounds	7	5.337
Non cyclic carbons ( below 30°C )	33	2.5323
Over all AAD%	40	3.93465

#### Table 4.3 The AAD% of Sato from experimental data

# 4.2 Experimental Data

The experimental data of the critical properties, obtained from literature for the purpose of this investigation consist of 108 values of pure components as shown in table 4-4. Table 4.5 shows the critical properties of polar components.

Chemical family	No. of compound	Data Reference
Organic		
Paraffins	70	21,22,25,28
Olefins	14	21,22,25,28
Diolefins	2	21,22,25,28
Acetylenes	2	21,22,25,28
Aromatics	8	21,22,25,28
Inorganic	12	21,22,25,28
Σ	108	

 Table 4-4 The pure component critical properties data

#### 4.3 Selection of the Correlations

Based on the findings of past investigators, of more recent correlations and on the criterion that a useful method must have input parameters which are available and accurate, the selection of correlations for each critical property was made. Several methods are available in the literatures for prediction of critical properties of pure substances. Most of these methods require the use of group contribution to determine the critical constants from molecular structures. It would be desirable to have an alternative easy method rather than the relatively complex method of group contribution and also to permit the estimation of the critical properties of substances for which group contribution are not available.

Certain methods were eliminated from evaluation because of their limited applicability or because they require input parameters which are not available or which are difficult to measure or predict accurately on their own right.

# 4.4 Development of the Correlation

The results indicate clearly that when evaluating critical volume ( $V_c$ ), and critical temperature ( $T_c$ ) there is no suitable equation that gives little deviation without using group contribution method. So in this work a simple equation was used for predicting critical temperature as a function of normal boiling point ( $T_b$ ) and molecular weight (Mwt) which are available for very large number of compounds in the literature, and with rare exception these data are very accurate. The proposed method uses only these accurate data.

The critical temperature generally can be estimated more accurately than other critical properties such as critical pressure and critical volume. So it is advisable to estimate critical temperature first.

# 4.5 Steps of Development

#### **Step One**

Since the critical temperature can be estimated more accurately than other critical properties (critical pressure and critical volume), so a new equation was proposed for prediction of critical temperature as a function of normal boiling point  $(T_b)$  and molecular weight (Mwt) as follows ;

 $Tc = -11.5565 - 1.03586Mwt + 2.075167Tb - 0.000281Mwt^{2}$  $- 0.00131Tb^{2} + 0.001827MwtTb$ 

(4.1)

This equation was obtained by fitting 30 point data of 30 components. The coefficient of correlation value of this fitting was 0.99999.

#### **Step Two**

A suitable value of critical pressure ( $P_C$ ) was assumed for a component and used in the following equations together with the critical temperature obtained in step one, to calculate critical volume ( $V_C$ )as in the following equations ;

$$Vc = \frac{R Tc}{R} [3.72 + 0.26 (\alpha - 7)]^{-1}$$
(3.1)

Where R is gas constant. And

$$\alpha = 0.9076 \left[ 1 + \frac{\left( Tb / Tc \right) \ln Pc}{1 - \left( Tb / Tc \right)} \right]$$
(3.2)

# **Step Three**

Critical pressure ( $P_C$ ) assumed in step two and critical temperature ( $T_C$ ) obtained in step one were used to calculate the acentric factor ( $\omega$ ). Using two methods:

#### Method one

Using the following equation of vapor pressure;

$$Log P_{VP} = A + \frac{B}{T}$$
(2.10)

By applying the value of vapor pressure and temperature at  $T_r = 0.7$  in equation (4.7), the acentric factor was calculated;

$$\omega = \frac{3}{7} \left[ \frac{\phi}{1 - \phi} \ LOG \ p_c \right] - 1 \tag{2.11}$$

Where

$$\phi = \frac{T_b}{T_C}$$

#### Method two

Using equation of Lee-Kesler to calculate the acentric factor ( $\omega$ )

$$\omega = \frac{\alpha}{\beta} \tag{2.12}$$

Where

$$\alpha = -\ln P_C - 5.97214 + 6.09648\phi^{-1} + 1.28862\ln\phi - 0.169347\phi^6$$
(2.13)

And

$$\beta = 15.2518 - 15.6875\phi^{-1} - 13.4721\ln\phi + 0.43577\phi^{6} \quad (2.14)$$

#### **Step Four**

To calculate critical pressure (P<sub>C</sub>) by

$$P_c = \frac{Z_c R T_c}{V_c} \tag{4.2}$$

Using critical compressibility factor ( $Z_C$ ) obtained for non – polar components from the following equation by Pitzer;

$$Z_c = 0291 - 0.08\omega \tag{3.30}$$

Or using critical compressibility factor  $(Z_C)$  of polar compounds by employing the equation of Wu and Stiel

$$Z_c = 0.2901 - 0.0879 \text{ w} - 0.00266 \text{ Y}$$
 (2.8)

This equation was modificated by statistical fitting and the following equation was obtained

$$Z_c = 0.2901 - 0.879^* \omega - 0.033583 \ Y \quad (4.3)$$

**Step Five** 

If the value of critical pressure ( $P_C$ ) calculated in step four is the same value as critical pressure ( $P_C$ ) assumed in step two, then this the correct critical pressure ( $P_C$ ) for the compound otherwise the value critical pressure ( $P_C$ ) obtained in step four would be assumed again in step two and the steps repeated until the value of critical pressure ( $P_C$ ) obtained in step four is the same as the value of critical pressure ( $P_C$ ) assumed with the tolerance of 10<sup>-4.</sup>

# 4.6 Application of the Developed Correlation

1. In order to clarify the application of the developed correlation an example is given below for Isobutane which was not included in statistical fitting:

 Table 4.5 Properties of Isobutane

comp	Mwt	<b>T</b> <sub>b</sub> ( <b>K</b> )	$T_{C}(K)$	<b>P</b> <sub>C</sub> (	$V_{\rm C}$ ( cm <sup>3</sup> /	Z <sub>C</sub>	ω
				bar )	mol)		
Isobutane	58.123	261.43	407.85	36.4	262.7	0.278	0.186

Assume  $P_C = 40$  bar

Table 4.6

Iteration	T <sub>C</sub>	V <sub>C</sub>	ω	Z <sub>C</sub>	P <sub>C</sub>
No.	( K)	( cm <sup>3</sup> / mol)			(bar)
1	407.867	229.83	0.1903	0.2721	39.998
2	407.867	240.82	0.189	0.2254	38.103
3	407.867	255.815	0.1885	0.2601	37.56
4	407.867	261.927	0.1865	0.2689	36.499
% dev.	0.00416	0.291	0.2688	0.327	0.0618

# Trial 2

1. By using the value of  $P_c$  estimated in step four, the new value of  $V_c$  is calculated in step two. These procedures are continued in the same way, and the new values of  $V_c$ ,  $Z_c$  and  $P_c$  are estimated.

In this fashion successive approximations are continued till P<sub>c</sub> converges.

2. In order to clarify the application of the developed correlation an example for polar component is given below for Ammonia.

Comp.	Mwt	T <sub>b</sub>	T <sub>C</sub>	P <sub>C</sub>	V <sub>C</sub>	Z <sub>C</sub>	ω
		(K)	(K)	(bar)	(cm <sup>3</sup> / mol)		
Ammonia	17.031	239.82	405.4	113.53	72.47	0.278	0.25

**Table 4.7 properties of Ammonia** 

Assume  $P_C = 40$  bar

Table 4.8

Iteration	T <sub>C</sub>	V <sub>C</sub>	ω	Z <sub>C</sub>	P <sub>C</sub>
No.	(K)	( cm <sup>3</sup> / mol)			(bar)
1	390.6	86.45	0.21	0.22	101.4
2	390.6	85.14	0.215	0.224	103
3	390.6	83.34	0.22	0.231	105.16
4	390.6	81.82	0.23	0.236	107.12
5	390.6	80.2	0.235	0.24	109.3013
6	390.6	78.45	0.237	0.2409	111.7327
7	390.6	76.586	0.24	0.243	114.45

# 4.7 Further Attempt to Improve The Method of Correlation for Prediction of Critical Properties

1. Changing the constants of equation for acentric factor by fitting 30 data points to improve the values of acentric factor for the components

$$Z_c = 0.2901 - 0.879 * \omega - 0.033583$$

- 2. As more accurate values of acentric factor predicted, the more accurate values of critical compressibility factor ( $Z_C$ ) and critical pressure predicted as shown in table 4.6.
- 3. Changing the constants of Wu and Stiel equation for critical compressibility factor ( $Z_C$ ) calculated by fitting 16 data point .The equation obtained is as follows;

$$Z_c = 0.2901 - 0.879 * \omega - 0.033583$$

Note that all the compounds that are predicted by this work are not included in statistical fitting of the developed equations of acentric factor and critical compressibility factor.

# **CHAPTER FIVE**

# Discussions

# **5.1 Introduction**

This chapter is indented to deal with the discussion of the results presented in chapter four. Many excellent estimation methods have been proposed in literature and mentioned in chapter three, For instance, Lydesen, Thodos, constantionu and Gani, Joback, Toshikatsu Hakuta and Mitsuho Hirata and others. However, these authors used group contribution estimation of critical properties of compounds. This method, that is group contribution method, is rather complicated and difficult because it deals with chemical structures and concerned with number of atoms and bonds between them. So for this reason a new and relatively simple, accurate and generalized method is proposed in this research which does not use group contribution technique.

In this work critical temperature, critical pressure, and critical volume can be estimated solely from data of normal boiling point and molecular weight of pure substance by means of successive approximations that are repeated until the calculated critical pressure converges.

As a result of this study the method discussed below was proposed as the best available generalized techniques for estimating the critical properties of pure components.

# **5.2 Experimental Data**

Appendix A shows the experimental values of critical properties of various compounds including Hydrocarbons, Aromatics, Amines, Alcohols and other polar compounds. These data were taken from Ref (20).

components	T <sub>Cexp</sub> .	V <sub>Cexp</sub> .	P <sub>Cexp</sub>	Z <sub>Cexp.</sub>		
<b>Organic Compou</b>	ndsHyd	rocarb	ons	_		
Isobutane	407.85	259	36.4	0.278		
2,2,4-	544.1	470.1	25.4	0.253		
Trimethylpentane						
Cyclohexane	553	307.8	40.0	0.234		
Benzene	562	260.3	48.6	0.26		
Dimethylamine	456.4	182	56.4	0.27		
Triethylamine	535	389	30.3	0.265		
Methyl alcohol	512.6	118	80.9	0.224		
Ethyl alcohol	513.9	176.1	61.4	0.24		
n. Propyl alcohol	536.8	219	51.7	0.253		
Aromatic						
C <sub>6</sub> H <sub>6</sub>	562.05	256	48.95	0.271		
C <sub>8</sub> H <sub>10</sub>	630.3	370	37.32	0.262		
Organic Compounds						
Ammonia	405.5	72.5	113.5	0.244		
Water	647.3	57.1	221.2	0.235		
Chlorine	324.7	80.9	83.1	0.249		

Table 5.1 Typical experimental critical properties of some polar andnon-polar components

# **5.3 Method of Proposed Correlation of Critical Properties**

# 5.3.1 Step One: Estimation of Critical Temperature

A statistical program was used to obtain a general equation to calculate the critical temperature by using normal boiling point and molecular weight. This equation is then used to calculate the critical temperatures of various compounds which were not used in the fitting. The results of estimation of various groups of compounds are shown in table 5.2

Group of comp.	No. of data point	AAD%
Paraffine	114	1.8703
olefine	14	0.3422
Diolefine	2	2.2515
Aromatic	8	2.57

 Table 5.2 shows The AAD% of various group of compounds

The detail deviation of individual components is shown in appendix B.

Comparison between the above proposed method for estimating critical temperature and Klincewicz equation used also the normal boiling point and molecular weight in deriving his equation below;

$$T_{C} = 50.2 - 0.16 Mwt + 1.41 T_{h}$$

Table 5.3 shows the results of predicting critical temperature  $T_C$  for some none – polar compounds using the proposed method and Klincewicz method. Table 5.3 shows the results of comparison of predicting critical temperature ( $T_C$ ) for 15 polar compounds using the proposed method and Klincewicz method.

Comp.	Tc exp.	Te	AAD%	Tc this	AAD%
Comp.	i comp.				
		Klincewicz.(K)		work(K)	
CH <sub>4</sub>	19056	180.506	5.27	190.5799	1.04*10 <sup>-2</sup>
C <sub>2</sub> H <sub>6</sub>	305.32	310.4	1.66	305.04	<b>8.8</b> *10 <sup>-2</sup>
C <sub>3</sub> H <sub>8</sub>	369.8	361.059	2.147	370.09	7.15*10 <sup>-2</sup>
C <sub>4</sub> H <sub>10</sub>	425	420.58	1.04	425.35	5.5*10 <sup>-2</sup>
C <sub>4</sub> H <sub>10</sub>	407.85	410.9	0.747	408.203	8.664*10 <sup>-2</sup>
C <sub>5</sub> H <sub>10</sub>	452.7	449.03	0.8106	452.5329	<b>3.691*10<sup>-2</sup></b>
C <sub>5</sub> H <sub>12</sub>	469.7	465	1.0006	469.535	<b>3.5</b> *10 <sup>-2</sup>
C <sub>5</sub> H <sub>12</sub>	433.75	430.2	0.818	437.6675	0.90317
C <sub>6</sub> H <sub>12</sub>	504	509.71	1.132	503.12	0.173
C <sub>6</sub> H <sub>12</sub>	495	489.7	1.07	493.334	0.336
C <sub>6</sub> H <sub>14</sub>	497.5	491.2	1.266	498.887	0.278
C7H14	572.19	565	1.256	568.19	0.697
C <sub>7</sub> H <sub>14</sub>	569.5	553.4	2.82	561.03	1.486
C7H16	520.4	515	1.037	526.05	1.087
C <sub>8</sub> H <sub>16</sub>	567	560.8	1.0934	564.017	0.526
C <sub>8</sub> H <sub>18</sub>	559.6	550	1.715	562.714	0.556
C <sub>9</sub> H <sub>20</sub>	<b>577.8</b>	590.08	2.12	586.95	1.584
$C_{10}H_{14}$	652	660.71	1.335	633.156	2.89
$C_{10}H_{18}$	703.6	710.02	0.912	691.445	1.727
$C_{11}H_{24}$	639	631.34	1.198	638.59	6.2*10 <sup>-2</sup>
$C_{14}H_{22}$	708	699.59	1.187	771.668	0.518
<b>Overall AAD%</b>			1.4702%		0.6599%

Table 5.3 The results of using the proposed equation in this work and thatof Klincewicz for some non-polar compounds

As shown in table 5.3 the overall average deviation for 20 non-polar compounds by Klincewicz is 1.4702 % while in this work is 0.6599 %, and for 114 non-polar compounds shown in appendix B is 0.9878 %.

Table 5.4 shows the results of comparison of predicting critical temperature for 16 polar compounds using the method of Klincewicz and the proposed method in this work.

	Polar		Tc e	exp.	Tc		AAD	%	Tc thi	S	AAD%
	compou	nd			Klincewicz.	(K)			work(I	K)	
	H2O	64	7.3		573.45	11	.408	6	521.563		3.976
	NH3	40	5.5		386.6	4	.66	3	99.014		1.599
-	HCL	32	4.7		375.916	1.	.577	, ,	320.89		1.1733
C	НЗОН	51	2.6		522.517	1.	9346	5	514.029		0.278
C2	2Н5ОН	51	3.9		539.594	4	.99	5	520.645		1.3125
Et	thylene	4	69		421.325	1	0.23	4	52.305		3.559
(	Oxide										
A	cetone	50	8.1		505.7105	0.4	4702	4	509.57		0.2893
I-P	ropanol	50	8.3		515.412	1.	.399		510.66		0.4642
N -P	Propanol	53	6.8		524.9	2	.21	4	531.97		0.899
C	H3CL	41	6.3		409.15	1.7	1775	4	417.76		0.35
(	CH3F	3	15		310.78	1.	3396	, ,	317.15		0.68
	<b>R</b> <sub>12</sub>	3	85		379.61	1.	3999	, ,	383.72		0.332
	R <sub>11</sub>	47	1.2		450.97	4.2	2932	4	60.029		2.3707
	R <sub>22</sub>	36	9.3		361.85	2.	.017	3	671.603		0.623
	R <sub>32</sub>	35	1.6		340.03	3.2	2906	3	49.016		0.736

Table 5.4 shows the comparison of results of using the proposed equation inthis work and that of Klincewicz for 15 polar compounds

R <sub>124</sub>	399.9	380.51	4.8487	387.908	2.998
Overall AAD%			3.523%	1.35	525%

As shown in table 5.4 the overall average deviation for 16 polar compounds is 3.523 % using Klincewicz method and 1.3525 % for predicting the critical temperatures for these compounds.

It is clear from tables 5.3 and 5.4 that the proposed method for estimating the critical temperature for non-polar and polar compounds is more accurate than that of Klincewicz method.

Comparing this method with group contribution method for predicting critical temperature is as follows. Ambarose (19) stated that AAD% for his method of group contribution for 400 pure non - polar compounds was 4.3% and Jobake (19) stated that AAD% for his method of group contribution for 400 non-polar pure compounds was 3.6%, the AAD % for prediction of critical temperature  $(T_c)$  for 114 non-polar compounds in this work was 0.9878%. It is clear that the prediction method in this work is better than that are mentioned above.

## 5.3.2 Step Two: Estimation of Critical Volume

The following equations of Reidel (2) were used to estimate the critical volume after assumption of suitable value for critical pressure;

$$V_c = \frac{RT_c}{P_c} [3.72 + 0.26(\alpha - 7.0)]^{-1}$$

Where

$$\alpha = 0.9076 \left[ 1.0 + \frac{\binom{T_b}{T_c} \ln P_c}{1.0 - \binom{T_b}{T_c}} \right]$$

Reidel pointed out that the above equations can be used only for hydrocarbons. However, in this work the above equations used for various groups of compounds including polar compounds.

After obtaining the most accurate value of critical pressure ( $P_c$ ), the results of the values of critical volumes ( $V_c$ ) using the above equations are shown in table 5.5 for various groups of compounds. Table 5.5 also includes the values of critical volumes for some compounds estimated by Viswanth.

Chemical family	No. of	AAD % Reidel	AAD %
	Data point	method	Viswanath
			method
Paraffins	52	1.7647	3.359
Olefins	14	1.7638	5.09029
Diolefins	2	0.995621	5.2822
Acetylenes	2	1.8204305	4.51556
Aromatics	8	1.3764	2.802511
Over all AAD%	78	1.7651%	3.8881%

 Table 5.5The AAD% of critical volume estimated by Reidel method and

 Viswanath method

The overall AAD% for predicting critical volumes by Reidel method which is used in this work is 1.7651% and the AAD% for predicting critical volumes by Viswanath method for the same compounds is 3.8881 %.

Comparing this method with group contribution method for critical volume is as follows: Amabrose stated that AAD% for his method of group contribution for 400 pure compounds was 8.5% and Jobake stated that AAD% that AAD% for his method of group contribution was 7.5%. It is clear that prediction method used in this work for critical volume ( $V_c$ ) is better than those mentioned above.

# 5.3.3 Step Three: Estimation of Acentric Factor

The following equations of vapor pressure equation (19,20), and that of Lee-Kesler(19,20) were used to estimate acentric factor after applying the assumption of suitable value for critical pressure;

$$\omega = \frac{3}{7} \left[ \frac{\phi}{1 - \phi} LOG p_c \right] - 1$$

Where

$$\phi = \frac{T_b}{T_C}$$

Or using Lee-Kesler equation

$$\omega = \frac{\alpha}{\beta}$$

Where

$$\alpha = -\ln P_c - 5.97214 + 6.09648\phi^{-1} + 1.28862\ln\phi - 0.169347\phi^6$$
$$\beta = 15.2518 - 15.6875\phi^{-1} - 13.4721\ln\phi + 0.43577\phi^6$$

Acentric factor is very widely used as a parameter which in some manner is supposed to measure the complexity of a molecule with respect to both the geometry and polarity.

#### 5.3.4 Step Four: Estimation of Critical Compressibility Factor

For non-polar compounds, critical compressibility factor  $(Z_C)$  was obtained by Pitzer equation;

$$Z_c = 0.0291 - 0.08 \quad \omega \tag{5.1}$$

Wu and Stiel (26) modified the Lee-Kesler scheme by incorporating the polar substance water as a third reference fluid. The compression factor is given in this extended schemes is given by;

$$Z = Z_{\circ} + \omega Z_1 + YZ_2 \tag{5.2}$$

Where Y is a forth parameter characteristic for polar compounds. When the above equation was applied at the critical point, the following equation was obtained;

$$Z_{\rm C} = 0.2901 - 0.0879 \ \omega - 0.0226 \ {\rm Y}$$
 (5.3)

Where the first two terms on the right hand side of the above equation is for nonpolar compounds (Pitzer equation for  $Z_C$ ). When using equation (5.3) for estimating critical compressibility factor ( $Z_C$ ) for 16 polar compounds it is found that the deviation of the values of  $Z_C$  from the values of  $Z_C$  obtained from literature were rather high. The Wu and Steil equation was modified by statistical method based on values of  $Z_C$  obtained from literature. The following modified equation was obtained;

$$Z_C = 0.2901 - 0.879 * \omega - 0.033583 * Y$$
(5.4)

The fourth parameter (Y) of equation (5.4) is determined from PVT data for the fluid. Tables 5.6 indicate the results of estimating  $Z_C$  by Wu and Steil equation and modified form of this equation that is done in this work. It is obvious that using the modified Wu and Steil equation gives more accurate values of  $Z_C$  which is the original equation.

Polar compound	Z <sub>C</sub> lit.	Z <sub>CWu &amp;steil</sub>	$\Delta Z_{C}$	$Z_C$ in this work	$\Delta Z_{C}$
H2O	0.235	0.267	0.032	0.256	0.021
NH3	0.244	0.273	0.029	0.265	0.021
HCL	0.249	0.277	0.028	0.271	0.022
СНЗОН	0.224	0.275	0.051	0.268	0.044
С2Н5ОН	0.24	0.285	0.045	0.0282	0.042
Ethylene Oxide	0.259	0.277	0.018	0.271	0.012
Acetone	0.232	0.275	0.043	0.268	0.036
I- Propanol	0.248	0.2889	0.0409	0.2883	0.0403
N - Propanol	0.253	0.2912	0.0382	0.2918	0.0388
CH3CL	0.259	0.283	0.014	0.279	0.01
CH3F	0.24	0.273	0.033	0.265	0.025
R <sub>12</sub>	0.28	0.2896	9.6E-3	0.2893	9.3E-3
R <sub>11</sub>	0.279	0.2897	0.0107	0.2895	0.0105
R <sub>22</sub>	0.268	0.2863	0.0187	0.2845	0.0165
R <sub>32</sub>	0.241	0.274	0.033	0.266	0.025
R <sub>124</sub>	0.273	0.29012	0.01712	0.29013	0.01713

Table 5.6 The Comparison of the results of estimating Z<sub>c</sub> by Wu and Steil equation and modified form

# 5.3.5 Step Five: Estimation of Critical Pressure

The results of the values of critical pressure prediction by this four steps procedure proved to be satisfactory. For 11 non-polar compounds (in Appendix C) the overall percent deviation is 1.98%. Table 5.7 shows typical results of 20 non-polar compounds.

Comp.	Pc exp.(bar)	Pc cal.(bar)	Dev.%
CH <sub>4</sub>	45.99	45.1	0.6305
C <sub>2</sub> H <sub>6</sub>	48.72	48.09	1.29
C <sub>3</sub> H <sub>8</sub>	42.48	42.3	0.423
C <sub>4</sub> H <sub>10</sub>	37.96	37.6	0.9483
C <sub>4</sub> H <sub>10</sub>	36.4	36.05	0.961
C <sub>5</sub> H <sub>12</sub>	33.7	33.1	1.7
C <sub>5</sub> H <sub>12</sub>	31.96	31.8	0.5
C <sub>6</sub> H <sub>14</sub>	30.1	29.98	0.398
C <sub>7</sub> H <sub>16</sub>	27.4	27.12	1.02
C <sub>8</sub> H <sub>18</sub>	24.9	24.35	2.208
C <sub>9</sub> H <sub>20</sub>	25.3	25.1	0.79
C <sub>10</sub> H <sub>22</sub>	21.1	20.89	0.99
C <sub>11</sub> H <sub>24</sub>	19.8	19.45	0.38

Table 5.7 Typical results of 16 non-polar compounds

C <sub>12</sub> H <sub>26</sub>	18.2	18	1.09
C <sub>13</sub> H <sub>28</sub>	16.8	16.02	4.6
C <sub>14</sub> H <sub>30</sub>	15.7	15.25	2.86
OverallAAD%			1.2349%

Table 5.8 shows the results of predicted critical pressure of 16 polar compounds. The overall average deviation is 1.4671% which are regarded very satisfactory.

Comp.	Pc exp.(bar)	Pc cal.(bar)	Dev.%
H <sub>2</sub> O	221.2	220.1	0.497
NH <sub>3</sub>	113.5	111.73	0.835
HCl	83.1	80.9	2.64
CH <sub>3</sub> OH	80.9	78.98	2.37
C <sub>2</sub> H <sub>5</sub> OH	61.4	60.5	1.46
Ethyl oxide	71.9	70.76	1.58
Acetone	47	45.9	2.2
I-propanol	42.5	41.8	1.66
n-propanol	51.7	50.87	1.5
CH <sub>3</sub> Cl	67	66.65	0.522

Table 5.8 Results of 16 polar compounds

CH <sub>3</sub> F	56	55.46	0.964
<b>R</b> <sub>12</sub>	41.4	40.3	2.65
<b>R</b> <sub>11</sub>	44.1	43.7	0.907
R <sub>22</sub>	49.7	48.45	2.51
<b>R</b> <sub>32</sub>	58.3	57.91	0.668
<b>R</b> <sub>124</sub>	37.2	37.01	0.5107
<b>Overall AAD%</b>			1.4671%

It is important to mention that the group contribution method to estimate critical pressure indicate the AAD% is 4.6 for 400 compounds by Ambarose and 5.2 by Jobake for 400 compounds also.

So, these four steps method to predict critical properties seems to be better than those of group contribution method.

#### **5.2 Conclusions**

1- A fairly accurate and easy method for estimation of critical temperature, critical volume and critical pressure from data of normal boiling point and molecular weight of pure substance only. This method consists of four steps and by means of successive approximation that are repeated until calculation of the critical pressure converges. This method proved to be more accurate than group contribution method.

2- A new general equation has been obtained to predict critical temperature. This equation is a relation of normal boiling point and molecular weight;

$$Tc = -11.5565 - 1.03586Mwt + 2.075167Tb - 0.000281Mwt^{2}$$
$$-0.00131Tb^{2} + 0.001827MwtTb$$

The AAD% for 114 pure compounds obtained from the above equation is 0.9878% and for 16 polar compounds is 1.3525 %.

3- Reidel equation for critical volume was used successfully for non-polar and polar compounds. The AAD% for critical volume in these four steps of successive approximation using Reidel equation is 1.7651 % for 78 data point.

4. Modification of Wu and Steil equation in the following form;

 $Z_c = 0.2901 - 0.879 * \omega - 0.033583$  Y

Proved to be more successful than their original equation for critical compressibility factor ( $Z_C$ ) for polar compounds.

5. The results of the values of critical pressure predicted by these four steps of successive approximation proved to be satisfactory.

The AAD % for 114 non-polar compounds is 1.89% and the AAD % for 16 polar compounds is 1.4671 %.

# **5.3 Recommendations**

For those who would carry feature studies on related subject, the following recommendations may be considered:

- Applying the developed correlation for further available experimental data

   polar and non polar.
- 2. Studying the possibility of applying this correlation for binary mixture with an acceptable accuracy.

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Component	Mwt	T <sub>C</sub>	P <sub>C</sub>	T <sub>b</sub>	ω	V <sub>C</sub>
•		(K)	(bar)	(K)		$( cm^3/mol)$
CH <sub>4</sub>	16.043	190.56	45.99	111.66	0.011	98.6
C <sub>2</sub> H <sub>6</sub>	30.07	305.32	48.72	184.55	0.099	145.5
C <sub>3</sub> H <sub>8</sub>	44.097	369.83	42.48	231.02	0.152	200
C <sub>4</sub> H <sub>10</sub>	58.123	425.12	37.96	272.66	0.2	255
C <sub>4</sub> H <sub>10</sub>	58.123	407.85	36.4	261.34	0.186	259
$C_{5}H_{12}$	72.15	469.7	33.7	309.22	0.252	311
$C_{5}H_{12}$	72.15	460.4	33.81	300.99	0.229	306
C <sub>5</sub> H <sub>12</sub>	72.15	433.8	31.96	282.65	0.197	307
C <sub>6</sub> H <sub>14</sub>	86.177	507.6	30.25	341.88	0.3	368
C <sub>6</sub> H <sub>14</sub>	86.177	497.5	30.1	333.4	0.278	366.7
C <sub>6</sub> H <sub>14</sub>	86.177	504.4	31.2	336.4	0.273	366.7
C <sub>6</sub> H <sub>14</sub>	86.177	<b>488.</b> 7	30.8	322.87	0.233	359.1
C <sub>6</sub> H <sub>14</sub>	86.177	4999.9	31.3	331.12	0.248	357
C <sub>7</sub> H <sub>16</sub>	100.204	540.2	27.4	371.57	0.35	428
C <sub>7</sub> H <sub>16</sub>	100.204	520.4	27.7	352.32	0.287	415.8
$C_{7}H_{16}$	100.204	519.7	27.4	353.62	0.304	417.5
$C_7H_{16}$	100.204	536.3	29.5	359.19	0.269	414.1
C <sub>7</sub> H <sub>16</sub>	100.204	537.3	29.1	362.91	0.297	393
$C_7H_{16}$	100.204	530.1	27.3	363.18	0.331	421
C <sub>7</sub> H <sub>16</sub>	100.204	535.2	28.1	365.0	0.323	404
C <sub>7</sub> H <sub>16</sub>	100.204	540.5	28.9	366.63	0.311	415.8
C <sub>7</sub> H <sub>16</sub>	100.204	531.1	29.5	354.01	0.25	398
C <sub>8</sub> H <sub>18</sub>	114.231	568.7	24.9	398.82	0.399	492
C <sub>8</sub> H <sub>18</sub>	114.231	567.8	28.7	379.44	0.248	488
C <sub>8</sub> H <sub>18</sub>	114.231	559.6	24.8	390.8	0.378	464
C <sub>8</sub> H <sub>18</sub>	114.231	563.6	25.5	392.08	0.378	471.1
C <sub>8</sub> H <sub>18</sub>	114.231	561.7	25.4	390.66	0.372	476
C <sub>9</sub> H <sub>20</sub>	128.258	577.8	23.5	405.97	0.383	525
C <sub>8</sub> H <sub>18</sub>	114.231	563.4	26.3	388.76	0.347	468.2
C <sub>8</sub> H <sub>18</sub>	114.231	553.5	25.6	382.58	0.344	472
C <sub>8</sub> H <sub>18</sub>	114.231	550	24.9	382.26	0.357	482

# Appendix A The Critical Properties of non-polar Components

**<u>Critical Properties of Paraffin Components</u>** 

C <sub>8</sub> H <sub>18</sub>	114.231	568.8	26.9	390.88	0.338	458.8
C <sub>8</sub> H <sub>18</sub>	114.231	563.4	72.3	382.99	0.298	436
C <sub>8</sub> H <sub>18</sub>	114.231	573.5	28.2	387.92	0.291	455.1
C <sub>8</sub> H <sub>18</sub>	114.231	566.3	27.3	386.62	0.316	456.2
C <sub>9</sub> H <sub>20</sub>	128.258	594.6	22.9	423.97	0.445	555
C <sub>9</sub> H <sub>20</sub>	128.258	<b>58</b> 7	23.1	416.44	0.423	529
C <sub>9</sub> H <sub>20</sub>	128.258	<b>592.7</b>	25.3	406.18	0.301	490
$C_{9}H_{20}$	128.258	574.6	24.9	395.44	0.314	504.0
$C_{10}H_{22}$	142.285	617.7	21.1	447.3	0.490	624
$C_{10} H_{22}$	142.285	609.6	23.2	428.83	0.383	564
$C_{11}H_{24}$	156.312	639.0	19.8	469.08	0.537	689.0
$C_{12}H_{26}$	170.338	658	18.2	489.48	0.576	754.0
$C_{13}H_{28}$	184.365	675.0	16.8	508.63	0.618	823
$C_{14}H_{30}$	198.392	693	15.7	526.76	0.644	894
$C_{15}H_{32}$	212.419	708	14.8	543.83	0.685	966
C <sub>16</sub> H <sub>34</sub>	226.446	723	14.0	559.98	0.718	1034
C <sub>17</sub> H <sub>36</sub>	240.473	736	13.4	574.56	0.753	1103

**<u>Critical Properties of Olefins Components</u>** 

Component	Mwt	T <sub>C</sub>	P <sub>C</sub>	T <sub>b</sub>	ω	V <sub>C</sub>
_		(K)	(bar)	(K)		( cm <sup>3</sup> /mol)
C <sub>3</sub> H <sub>4</sub>	42.081	398.25	55.75	240.34	0.13	162.8
C <sub>4</sub> H <sub>8</sub>	56.108	419.5	40.2	266.92	0.194	240.8
C <sub>4</sub> H <sub>8</sub>	56.108	435.5	42.1	276.87	0.203	233.8
C <sub>4</sub> H <sub>8</sub>	56.108	428.6	41.0	274.03	0.218	237.7
C <sub>5</sub> H <sub>10</sub>	70.143	464.8	35.6	303.11	0.237	298.4
C <sub>5</sub> H <sub>10</sub>	70.134	475.0	36.9	310.07	0.253	302.1
C <sub>5</sub> H <sub>10</sub>	70.134	470	38.6	311.7	0.336	292
C <sub>6</sub> H <sub>12</sub>	84.161	504	31.43	336.63	0.281	355.1
C <sub>7</sub> H <sub>14</sub>	98.188	537.3	29.2	366.79	0.343	409
C <sub>7</sub> H <sub>16</sub>	100.204	537.3	29.1	362.91	0.297	393
C <sub>8</sub> H <sub>16</sub>	112.215	567	26.8	394.44	0.266	468
$C_9H_{18}$	126.242	594	23.3	420	0.411	526
Component	Mwt	T <sub>C</sub>	P <sub>C</sub>	T <sub>b</sub>	ω	V <sub>C</sub>
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		<b>(K)</b>	(bar)	(K)		( cm <sup>3</sup> /mol)
C <sub>3</sub> H <sub>4</sub>	40.065	394	52.5	238.77	0.122	173.9
C <sub>4</sub> H <sub>6</sub>	54.092	425	43.2	268.62	0.195	221

#### **<u>Critical Properties of Diolefins Components</u>**

#### **<u>Critical Properties of Acetylenes Components</u>**

Component	Mwt	T <sub>C</sub>	P <sub>C</sub>	T <sub>b</sub>	ω	V <sub>C</sub>
_		(K)	(bar)	(K)		( cm <sup>3</sup> /mol)
$C_2H_2$	26.038	308.3	61.14	188.4	0.189	112.2
C <sub>4</sub> H <sub>8</sub>	56.108	419.5	40.2	266.92	0.194	240.8

#### **<u>Critical Properties of Aromatics Components</u>**

Component	Mwt	T <sub>C</sub>	P <sub>C</sub>	T <sub>b</sub>	ω	V <sub>C</sub>
		(K)	(bar)	(K)		( cm <sup>3</sup> /mol)
C <sub>6</sub> H <sub>6</sub>	78.114	562.05	48.95	353.24	0.21	256
$C_7H_8$	92.141	591.75	41.08	383.79	0.264	316
C <sub>8</sub> H <sub>10</sub>	106.167	630.3	37.32	417.59	0.312	370
C <sub>8</sub> H <sub>10</sub>	106.167	617	35.41	412.34	0.327	375
C <sub>8</sub> H <sub>10</sub>	106.167	616.2	35.11	411.53	0.322	378
C <sub>9</sub> H <sub>12</sub>	120.194	664.5	34.54	449.23	0.367	435
C <sub>9</sub> H <sub>12</sub>	120.194	649.1	32.32	442.49	0.377	435
C <sub>9</sub> H <sub>12</sub>	120.194	637.3	31.27	437.9	0.399	430

Polar compound	T <sub>b</sub>	T <sub>C</sub>	P <sub>C</sub>	V <sub>C</sub>	Z <sub>C</sub>	ω	Y
	(K)	(K)	(bar)	( cm <sup>3</sup> /mol)			
H2O	373.2	647.3	221.2	57.1	0.235	0.344	1.0
NH3	239.8	405.5	113.5	72.5	0.244	0.25	0.739
HCL	188.1	324.7	83.1	80.9	0.249	0.133	0.565
СНЗОН	337.7	512.6	80.9	118.0	0.224	0.556	0.652
С2Н5ОН	351.4	513.9	61.4	167.1	0.24	0.644	0.213
Ethylene Oxide	283.7	469	71.9	140.0	0.259	0.202	0.565
Acetone	329.2	508.1	47.0	209	0.232	0.304	0.652
I- Propanol	355.4	508.3	42.5	220	0.248	0.248	0.052
N - Propanol	370.3	536.8	51.7	219	0.253	0.254	-0.052
CH3CL	249.1	416.3	67.0	138.9	0.259	0.153	0.304
CH3F	194.7	315	56	113.2	0.24	0.187	0.739
R <sub>12</sub>	245.2	385	41.4	216.7	0.28	0.204	0.021
R <sub>11</sub>	296.9	471.2	44.1	247.8	0.279	0.189	0.015
R <sub>22</sub>	232.4	369.3	49.7	165.6	0.268	0.221	0.164
R <sub>32</sub>	221.5	351.6	58.3	120.8	0.241	0.271	0.704
<b>R</b> <sub>124</sub>	263	399.9	37.2	244	0.273	0.281	-0.001

## The Critical Properties of Polar Components

**Critical Properties of Polar Components** 

Component	$T_{C}(\mathbf{K})$	$T_{\rm C}$ cal. (K)	Dev. %
Ĩ		Ç ( )	
$CH_4$	190.56	190.5799	1.043357*10 <sup>-2</sup>
$C_2H_4$	282.38	278.4239	1.4009
C <sub>2</sub> H <sub>6</sub>	305.32	305.0491	8.872807*10 <sup>-2</sup>
$C_3H_4$	394	397.6411	0.9241482
C <sub>3</sub> H <sub>6</sub>	364.9	363.7002	0.3288104
C <sub>3</sub> H <sub>8</sub>	369.81	370.0946	7.154299*10 <sup>-2</sup>
$C_4H_6$	440	435.8007	0.9543957
C <sub>4</sub> H <sub>6</sub>	425	422.0742	0.6884191
C <sub>4</sub> H <sub>8</sub>	460	452.8507	1.554187
$C_4H_8$	419.5	415.7707	0.412231
$C_4H_8$	435.5	433.5311	0.4521016
$C_4H_8$	417.9	415.4891	0.5768983
C <sub>4</sub> H <sub>10</sub>	425.12	425.3574	5.583499*10 <sup>-2</sup>
C <sub>4</sub> H <sub>10</sub>	407.85	408.2034	8.664044*10 <sup>-2</sup>
C <sub>5</sub> H <sub>8</sub>	470	473.1609	0.6725425
C5H10	464.8	463.2576	0.3318458
C <sub>5</sub> H <sub>10</sub>	475	471.5228	0.7320428
C <sub>5</sub> H <sub>10</sub>	470	466.5704	0.7297078
C5H10	452.7	452.5329	3.691501*10 <sup>-2</sup>
C5H12	469.7	469.5349	3.515661*10 <sup>-2</sup>
C5H12	460.39	460.751	7.841676*10 <sup>-2</sup>
C <sub>5</sub> H <sub>12</sub>	433.75	437.6675	0.9031726
C <sub>6</sub> H <sub>12</sub>	504	503.1235	0.1739078
C <sub>6</sub> H <sub>12</sub>	495	493.3348	0.3364085
C <sub>6</sub> H <sub>14</sub>	507.6	507.357	4.788061*10 <sup>-2</sup>
C <sub>6</sub> H <sub>14</sub>	497.5	498.887	0.2787988
C <sub>6</sub> H <sub>14</sub>	504.4	504.1355	5.243167*10 <sup>-2</sup>
C <sub>6</sub> H <sub>14</sub>	488.7	493.0939	0.8990977
C <sub>6</sub> H <sub>14</sub>	499.9	501.7596	0.371986
$C_7\overline{H_{14}}$	572.19	568.1995	0.6974044
C <sub>7</sub> H <sub>14</sub>	553	549.2634	0.6757022

#### Appendix B Results of Investigation for Critical Temperatures

**<u>Critical Temperatures for Organic Components</u>** 

$\begin{array}{c ccccc} C_7H_{14} & 537.3 & 534.6062 & 0.5013562 \\ C_7H_{16} & 540.2 & 539.8822 & 5.88392*10^{-2} \\ C_7H_{16} & 530.1 & 532.0615 & 0.3700336 \\ C_7H_{16} & 535.2 & 535.5968 & 7.413855^{*}10^{-2} \\ C_7H_{16} & 537.3 & 537.7984 & 9.276254^{*}10^{-2} \\ C_7H_{16} & 519.7 & 524.3073 & 0.8865189 \\ C_7H_{16} & 536.3 & 539.5013 & 0.5969222 \\ C_8H_{16} & 567 & 564.0172 & 0.5260649 \\ C_8H_{18} & 568.7 & 568.6844 & 2.747494^{*}10^{-3} \\ C_8H_{18} & 559.6 & 562.7146 & 0.5565804 \\ C_8H_{18} & 559.6 & 562.7146 & 0.5565804 \\ C_8H_{18} & 559.6 & 562.7146 & 0.5565804 \\ C_8H_{18} & 555.4 & 566.0217 & 0.1099475 \\ C_8H_{18} & 555.5 & 559.0392 & 1.000767 \\ C_8H_{18} & 555.5 & 559.0392 & 1.000767 \\ C_8H_{18} & 555.5 & 555.4148 & 1.166326 \\ C_8H_{18} & 5562 & 566.9741 & 0.8850749 \\ C_8H_{18} & 567 & 568.701 & 0.299992 \\ C_8H_{18} & 567 & 568.701 & 0.299992 \\ C_8H_{18} & 567 & 568.701 & 0.299992 \\ C_8H_{18} & 566.3 & 569.5404 & 0.1301719 \\ C_8H_{18} & 567 & 568.701 & 0.299992 \\ C_8H_{18} & 566.3 & 569.6481 & 0.5912316 \\ C_9H_{18} & 576.5 & 577.945 & 0.2506517 \\ C_8H_{18} & 573.5 & 577.971 & 0.7492797 \\ C_8H_{18} & 566.3 & 569.6481 & 0.5912316 \\ C_9H_{20} & 594.6 & 594.579 & 3.520864*10^{-3} \\ C_9H_{20} & 587 & 590.4653 & 0.5903462 \\ C_9H_{20} & 5877 & 590.4653 & 0.5903462 \\ C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ C_9H_{20} & 607.6 & 615.1904 & 1.249251 \\ C_{14}H_{22} & 708 & 771.6688 & 0.5181867 \\ C_{14}H_{22} $				
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>7</sub> H <sub>14</sub>	537.3	534.6062	0.5013562
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>7</sub> H <sub>16</sub>	540.2	539.8822	5.88392*10 <sup>-2</sup>
$\begin{array}{c ccccc} C_7H_{16} & 535.2 & 535.5968 & 7.413855^{*}10^2 \\ C_7H_{16} & 540.5 & 539.7595 & 0.1370101 \\ C_7H_{16} & 537.3 & 537.7984 & 9.276254^{*}10^2 \\ C_7H_{16} & 519.7 & 524.3073 & 0.8865189 \\ C_7H_{16} & 536.3 & 539.5013 & 0.5969222 \\ C_8H_{16} & 567 & 564.0172 & 0.5260649 \\ C_8H_{18} & 568.7 & 568.6844 & 2.747494^{*}10^3 \\ C_8H_{18} & 559.6 & 562.7146 & 0.5565804 \\ C_8H_{18} & 563.6 & 565.0589 & 0.2588579 \\ C_8H_{18} & 565.4 & 566.0217 & 0.1099475 \\ C_8H_{18} & 555.4 & 566.0217 & 0.1099475 \\ C_8H_{18} & 555.5 & 559.0392 & 1.000767 \\ C_8H_{18} & 555.4 & 566.0217 & 0.1099475 \\ C_8H_{18} & 555.4 & 566.0217 & 0.1099475 \\ C_8H_{18} & 555.5 & 559.0392 & 1.000767 \\ C_8H_{18} & 550 & 556.4148 & 1.166326 \\ C_8H_{18} & 550 & 556.4148 & 1.166326 \\ C_8H_{18} & 562 & 566.9741 & 0.8850749 \\ C_8H_{18} & 567 & 568.701 & 0.299992 \\ C_8H_{18} & 567.5 & 577.945 & 0.2506517 \\ C_8H_{18} & 566.3 & 569.6481 & 0.5912316 \\ C_9H_{18} & 573.5 & 577.971 & 0.7492797 \\ C_8H_{18} & 566.3 & 569.6481 & 0.5912316 \\ C_9H_{18} & 594 & 593.8601 & 2.35509^{*}10^{-2} \\ C_9H_{20} & 587 & 590.4653 & 0.5903462 \\ C_9H_{20} & 587 & 590.4653 & 0.5903462 \\ C_9H_{20} & 587 & 590.4653 & 0.5903462 \\ C_9H_{20} & 577.8 & 586.956 & 1.584632 \\ C_9H_{20} & 577.8 & 586.956 & 1.584632 \\ C_10H_{22} & 607.6 & 615.1904 & 1.249252 \\ C_9H_{20} & 607.1 & 614.8072 & 1.269513 \\ C_{10}H_{12} & 609.6 & 622.8995 & 2.181677 \\ C_{11}H_{24} & 639 & 638.5988 & 6.278311^{*}10^{-2} \\ C_{12}H_{26} & 658 & 658.0058 & 8.812067^{*}10^{-4} \\ C_{13}H_{28} & 675 & 675.3093 & 0.0458261 \\ C_{14}H_{22} & 708 & 771.6688 & 0.5181867 \\ C_{14}H_{22} & 708 & 771.6688 & 0.5181867 \\ C_{14}H_{22} & 708 & 771.6653 & 4.727639^{*}10^{-2} \\ \end{array}$	C <sub>7</sub> H <sub>16</sub>	530.1	532.0615	0.3700336
$\begin{array}{c c} C_7 H_{16} & 540.5 & 539.7595 & 0.1370101 \\ C_7 H_{16} & 537.3 & 537.7984 & 9.276254*10^2 \\ C_7 H_{16} & 519.7 & 524.3073 & 0.8865189 \\ C_7 H_{16} & 536.3 & 539.5013 & 0.5969222 \\ C_8 H_{16} & 567 & 564.0172 & 0.5260649 \\ C_8 H_{18} & 568.7 & 568.6844 & 2.747494*10^3 \\ C_8 H_{18} & 559.6 & 562.7146 & 0.5565804 \\ C_8 H_{18} & 563.6 & 565.0589 & 0.2588579 \\ C_8 H_{18} & 563.4 & 565.0589 & 0.2588579 \\ C_8 H_{18} & 563.4 & 565.4199 & 0.3585192 \\ C_8 H_{18} & 550 & 556.4148 & 1.166326 \\ C_8 H_{18} & 550 & 556.4148 & 1.166326 \\ C_8 H_{18} & 562 & 566.9741 & 0.8850749 \\ C_8 H_{18} & 562 & 566.9741 & 0.8850749 \\ C_8 H_{18} & 567 & 568.701 & 0.299992 \\ C_8 H_{18} & 567 & 568.701 & 0.299992 \\ C_8 H_{18} & 567 & 568.701 & 0.299992 \\ C_8 H_{18} & 576.5 & 577.945 & 0.2506517 \\ C_8 H_{18} & 576.5 & 577.945 & 0.2506517 \\ C_8 H_{18} & 566.3 & 569.6481 & 0.5912316 \\ C_9 H_{18} & 594 & 593.8601 & 2.35509*10^{-2} \\ C_9 H_{20} & 587 & 590.4653 & 0.5903462 \\ C_9 H_{20} & 587 & 590.4653 & 0.5903462 \\ C_9 H_{20} & 587 & 590.4653 & 0.5903462 \\ C_9 H_{20} & 577.8 & 586.956 & 1.584632 \\ C_9 H_{20} & 577.8 & 586.956 & 1.584632 \\ C_9 H_{20} & 607.6 & 615.1904 & 1.249251 \\ C_1 H_{22} & 6$	C <sub>7</sub> H <sub>16</sub>	535.2	535.5968	7.413855*10 <sup>-2</sup>
$\begin{array}{cccccc} C_7 H_{16} & 537.3 & 537.7984 & 9.276254*10^2 \\ C_7 H_{16} & 519.7 & 524.3073 & 0.8865189 \\ C_7 H_{16} & 536.3 & 539.5013 & 0.5969222 \\ C_8 H_{16} & 567 & 564.0172 & 0.5260649 \\ C_8 H_{18} & 568.7 & 568.6844 & 2.747494*10^3 \\ C_8 H_{18} & 559.6 & 562.7146 & 0.5565804 \\ C_8 H_{18} & 563.6 & 565.0589 & 0.2588579 \\ C_8 H_{18} & 563.4 & 565.0589 & 0.2588579 \\ C_8 H_{18} & 563.4 & 565.4199 & 0.3585192 \\ C_8 H_{18} & 550 & 556.4148 & 1.166326 \\ C_8 H_{18} & 550 & 556.4148 & 1.166326 \\ C_8 H_{18} & 562 & 566.9741 & 0.8850749 \\ C_8 H_{18} & 562 & 566.9741 & 0.8850749 \\ C_8 H_{18} & 567 & 568.701 & 0.299992 \\ C_8 H_{18} & 567 & 568.701 & 0.299992 \\ C_8 H_{18} & 567 & 568.701 & 0.299992 \\ C_8 H_{18} & 576.5 & 577.945 & 0.2506517 \\ C_8 H_{18} & 566.3 & 569.6481 & 0.5912316 \\ C_9 H_{18} & 594 & 593.8601 & 2.35509*10^{-2} \\ C_9 H_{20} & 594.6 & 594.579 & 3.520864*10^{-3} \\ C_9 H_{20} & 587 & 590.4653 & 0.5903462 \\ C_9 H_{20} & 577.8 & 586.956 & 1.584632 \\ C_9 H_{20} & 607.1 & 614.8072 & 1.269513 \\ C_1 H_{22} & 607.6 & 615.1904 & 1.249252 \\ C_9 H_{20} & 607.6 & 615.1904 & 1.249252 \\ C_9 H_{20} & 577.8 & 586.956 & 1.584632 \\ C_9 H_{20} & 607.1 & 614.8072 & 1.269513 \\ C_1 H_{22} & 609.6 & 622.8995 & 2.181677 \\ C_{11} H_{24} & 639 & 638.5988 & 6.278311*10^2 \\ C_{12} H_{26} & 658 & 658.0058 & 8.812067*10^4 \\ C_{13} H_{28} & 675 & 675.3093 & 0.0453261 \\ C_{14} H_{22} & 708 & 771.6688 & 0.5181867 \\ C_{14} H_{30} & 693 & 693.1696 & 2.447571*10^2 \\ C_{15} H_{32} & 708 & 707.6653 & 4.727639*10^2 \\ \end{array}$	C <sub>7</sub> H <sub>16</sub>	540.5	539.7595	0.1370101
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>7</sub> H <sub>16</sub>	537.3	537.7984	9.276254*10 <sup>-2</sup>
$\begin{array}{c cccc} C_7H_{16} & 536.3 & 539.5013 & 0.5969222 \\ C_8H_{16} & 567 & 564.0172 & 0.5260649 \\ C_8H_{18} & 568.7 & 568.6844 & 2.747494*10^3 \\ C_8H_{18} & 559.6 & 562.7146 & 0.5565804 \\ C_8H_{18} & 563.4 & 565.0589 & 0.2588579 \\ C_8H_{18} & 563.4 & 565.0217 & 0.1099475 \\ C_8H_{18} & 563.4 & 565.4199 & 0.3585192 \\ C_8H_{18} & 553.5 & 559.0392 & 1.000767 \\ C_8H_{18} & 550 & 556.4148 & 1.166326 \\ C_8H_{18} & 562 & 566.9741 & 0.8850749 \\ C_8H_{18} & 562 & 566.9741 & 0.8850749 \\ C_8H_{18} & 567 & 568.701 & 0.299992 \\ C_8H_{18} & 567 & 568.701 & 0.299992 \\ C_8H_{18} & 576.5 & 577.945 & 0.2506517 \\ C_8H_{18} & 576.5 & 577.945 & 0.2506517 \\ C_8H_{18} & 576.5 & 577.971 & 0.7492797 \\ C_8H_{18} & 566.3 & 569.6481 & 0.5912316 \\ C_9H_{18} & 594 & 593.8601 & 2.35509*10^{-2} \\ C_9H_{20} & 594.6 & 594.579 & 3.520864*10^{-3} \\ C_9H_{20} & 597.8 & 586.956 & 1.584632 \\ C_9H_{20} & 577.8 & 586.956 & 1.584632 \\ C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ C_9H_{20} & 607.1 & 614.8072 & 1.269513 \\ C_{10}H_{22} & 607.6 & 615.1904 & 1.249252 \\ C_{9}H_{20} & 607.6 & 622.8995 & 2.181677 \\ C_{11}H_{24} & 639 & 638.5988 & 6.278311*10^{-2} \\ C_{12}H_{26} & 658 & 658.0058 & 8.812067*10^{-4} \\ C_{13}H_{28} & 675 & 675.3093 & 0.0458261 \\ C_{14}H_{30} & 693 & 693.1696 & 2.447571*10^{-2} \\ C_{15}H_{32} & 708 & 771.6688 & 0.5181867 \\ C_{14}H_{30} & 693 & 693.1696 & 2.447571*10^{-2} \\ \end{array}$	C <sub>7</sub> H <sub>16</sub>	519.7	524.3073	0.8865189
$\begin{array}{c cccc} C_8H_{16} & 567 & 564.0172 & 0.5260649 \\ \hline C_8H_{18} & 568.7 & 568.6844 & 2.747494*10^3 \\ \hline C_8H_{18} & 559.6 & 562.7146 & 0.5565804 \\ \hline C_8H_{18} & 563.6 & 565.0589 & 0.2588579 \\ \hline C_8H_{18} & 565.4 & 566.0217 & 0.1099475 \\ \hline C_8H_{18} & 563.4 & 565.4199 & 0.3585192 \\ \hline C_8H_{18} & 553.5 & 559.0392 & 1.000767 \\ \hline C_8H_{18} & 550 & 556.4148 & 1.166326 \\ \hline C_8H_{18} & 562 & 566.9741 & 0.8850749 \\ \hline C_8H_{18} & 567 & 568.701 & 0.299992 \\ \hline C_8H_{18} & 567 & 568.701 & 0.299992 \\ \hline C_8H_{18} & 576.5 & 577.945 & 0.2506517 \\ \hline C_8H_{18} & 576.5 & 577.945 & 0.2506517 \\ \hline C_8H_{18} & 566.3 & 569.6481 & 0.5912316 \\ \hline C_9H_{18} & 594 & 593.8601 & 2.35509*10^{-2} \\ \hline C_9H_{20} & 587 & 590.4653 & 0.5903462 \\ \hline C_9H_{20} & 594.6 & 594.579 & 3.520864*10^{-3} \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.6 & 622.8995 & 2.181677 \\ \hline C_{11}H_{24} & 639 & 638.5988 & 6.278311*10^2 \\ \hline C_{12}H_{26} & 658 & 658.0058 & 8.812067*10^4 \\ \hline C_{13}H_{28} & 675 & 675.3093 & 0.0458261 \\ \hline C_{14}H_{30} & 693 & 693.1696 & 2.447571*10^2 \\ \hline C_{15}H_{32} & 708 & 771.6688 & 0.5181867 \\ \hline C_{14}H_{30} & 693 & 693.1696 & 2.447571*10^2 \\ \hline \end{array}$	C <sub>7</sub> H <sub>16</sub>	536.3	539.5013	0.5969222
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>8</sub> H <sub>16</sub>	567	564.0172	0.5260649
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>8</sub> H <sub>18</sub>	568.7	568.6844	2.747494*10 <sup>-3</sup>
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>8</sub> H <sub>18</sub>	559.6	562.7146	0.5565804
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	563.6	565.0589	0.2588579
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	565.4	566.0217	0.1099475
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>8</sub> H <sub>18</sub>	563.4	565.4199	0.3585192
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	553.5	559.0392	1.000767
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	550	556.4148	1.166326
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	562	566.9741	0.8850749
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	568.8	569.5404	0.1301719
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	567	568.701	0.299992
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	576.5	577.945	0.2506517
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	563.4	570.0463	1.179666
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	C <sub>8</sub> H <sub>18</sub>	573.5	577.7971	0.7492797
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>8</sub> H <sub>18</sub>	566.3	569.6481	0.5912316
$\begin{array}{c cccc} C_9H_{20} & 594.6 & 594.579 & 3.520864*10^{-3} \\ \hline C_9H_{20} & 587 & 590.4653 & 0.5903462 \\ \hline C_9H_{20} & 577.8 & 586.956 & 1.584632 \\ \hline C_9H_{20} & 607.6 & 615.1904 & 1.249252 \\ \hline C_9H_{20} & 607.1 & 614.8072 & 1.269513 \\ \hline C_{10}H_{18} & 703.6 & 691.4445 & 1.727618 \\ \hline C_{10}H_{22} & 617.7 & 617.8368 & 0.0221434 \\ \hline C_{10}H_{22} & 609.6 & 622.8995 & 2.181677 \\ \hline C_{11}H_{24} & 639 & 638.5988 & 6.278311*10^{-2} \\ \hline C_{12}H_{26} & 658 & 658.0058 & 8.812067*10^{-4} \\ \hline C_{13}H_{28} & 675 & 675.3093 & 0.0458261 \\ \hline C_{14}H_{22} & 708 & 771.6688 & 0.5181867 \\ \hline C_{15}H_{32} & 708 & 707.6653 & 4.727639*10^{-2} \end{array}$	C <sub>9</sub> H <sub>18</sub>	594	593.8601	2.35509*10 <sup>-2</sup>
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>9</sub> H <sub>20</sub>	594.6	594.579	3.520864*10 <sup>-3</sup>
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>9</sub> H <sub>20</sub>	587	590.4653	0.5903462
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>9</sub> H <sub>20</sub>	577.8	586.956	1.584632
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>9</sub> H <sub>20</sub>	607.6	615.1904	1.249252
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>9</sub> H <sub>20</sub>	607.1	614.8072	1.269513
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	C <sub>10</sub> H <sub>18</sub>	703.6	691.4445	1.727618
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	$C_{10}H_{22}$	617.7	617.8368	0.0221434
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	$C_{10}H_{22}$	609.6	622.8995	2.181677
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	$C_{11}H_{24}$	639	638.5988	<b>6.278311*10</b> <sup>-2</sup>
$\begin{tabular}{ c c c c c c c c c c c c c c c c c c c$	$C_{12}H_{26}$	658	658.0058	8.812067*10 <sup>-4</sup>
$\begin{array}{ c c c c c c c c }\hline C_{14}H_{22} & 708 & 771.6688 & 0.5181867 \\\hline C_{14}H_{30} & 693 & 693.1696 & 2.447571^{*}10^{-2} \\\hline C_{15}H_{32} & 708 & 707.6653 & 4.727639^{*}10^{-2} \\\hline \end{array}$	C <sub>13</sub> H <sub>28</sub>	675	675.3093	0.0458261
$\begin{array}{ c c c c c c c c }\hline \hline C_{14}H_{30} & \hline 693 & 693.1696 & 2.447571*10^{-2} \\ \hline C_{15}H_{32} & 708 & 707.6653 & 4.727639*10^{-2} \\ \hline \end{array}$	$C_{14}H_{22}$	708	771.6688	0.5181867
C <sub>15</sub> H <sub>32</sub> 708 707.6653 4.727639*10 <sup>-2</sup>	C <sub>14</sub> H <sub>30</sub>	693	693.1696	2.447571*10 <sup>-2</sup>
	C <sub>15</sub> H <sub>32</sub>	708	707.6653	<b>4.727639*10</b> <sup>-2</sup>

$C_{16}H_{34}$	723	722	8.887664*10 <sup>-2</sup>
$C_{17}H_{36}$	736	736.2701	<b>3.670402*10<sup>-2</sup></b>
$C_{18}H_{38}$	747	746.485	6.893623*10 <sup>-2</sup>

## For Paraffin

Component	V <sub>C</sub>	<b>Dev. %</b>
	cm <sup>3</sup> /mol	
CH <sub>4</sub>	98.6	2.328188
$C_2H_6$	145.5	1.238974
$C_3H_8$	200	0.3022537
$C_{4}H_{10}$	255	0.5210727
C <sub>4</sub> H <sub>10</sub>	259	1.428069
$C_{5}H_{12}$	311	0.2459363
$C_{5}H_{12}$	306	0.1267477
$C_{5}H_{12}$	307	0.3065376
$C_{6}H_{14}$	368	0.1200634
C <sub>6</sub> H <sub>14</sub>	366.7	0.548376
C <sub>6</sub> H <sub>14</sub>	359.1	0.9970265
C <sub>6</sub> H <sub>14</sub>	357	0.2827535
$C_{7}H_{16}$	428	0.7105319
$C_{7}H_{16}$	415.8	0.5877606
C <sub>7</sub> H <sub>16</sub>	417.5	0.5811424
C <sub>7</sub> H <sub>16</sub>	393	3.033773
C <sub>7</sub> H <sub>16</sub>	421	2.553763*10 <sup>-2</sup>
C <sub>7</sub> H <sub>16</sub>	404	2.502222
C <sub>7</sub> H <sub>16</sub>	415.8	1.837983
C <sub>7</sub> H <sub>16</sub>	398	0.7638289
C <sub>8</sub> H <sub>18</sub>	492	1.463305
C <sub>8</sub> H <sub>18</sub>	471.1	0.4575564
C <sub>8</sub> H <sub>18</sub>	476	0.4303107
C <sub>9</sub> H <sub>20</sub>	525	5.66638*10 <sup>-2</sup>
C <sub>8</sub> H <sub>18</sub>	468.2	1.274307
C <sub>8</sub> H <sub>18</sub>	472	1.041865
C <sub>8</sub> H <sub>18</sub>	482	1.413116
C <sub>8</sub> H <sub>18</sub>	458.8	0.2808775

C <sub>8</sub> H <sub>18</sub>	455.1	1.78992
C <sub>8</sub> H <sub>18</sub>	456.2	0.9034728
$C_{9}H_{20}$	529	1.239948
C <sub>9</sub> H <sub>20</sub>	504.0	9.349035*10 <sup>-2</sup>
$C_{10}H_{22}$	564	0.577529

#### **For Olefins**

Component	Vc	Dev. %
••• <b>-</b> • <b>-</b> •••••	cm <sup>3</sup> /mol	
$C_3H_4$	162.8	1.90419
$C_4H_8$	240.8	1.576471
C <sub>4</sub> H <sub>8</sub>	233.8	6.257539*10 <sup>-2</sup>
$C_4H_8$	237.7	0.8948185
C <sub>5</sub> H <sub>10</sub>	298.4	2.109456
$C_{6}H_{12}$	355.1	0.4697008
$C_{7}H_{14}$	409	2.810162
$C_{7}H_{16}$	393	3.033773
C <sub>8</sub> H <sub>16</sub>	468	2.55518
C <sub>9</sub> H <sub>18</sub>	526	1.318129

#### **For Diolefins**

Component	V <sub>C</sub>	Dev. %
	cm <sup>3</sup> /mol	
C <sub>3</sub> H <sub>4</sub>	173.9	0.405582
$C_4H_6$	221	1.058566

## **For Acetylenes**

Component	V <sub>C</sub>	Dev. %
_	cm <sup>3</sup> /mol	
$C_2H_2$	112.2	2.06439
$C_4H_8$	240.8	1.576471

Component	V <sub>C</sub>	Dev. %
	cm³/mol	
$C_6H_6$	256	1.08453
$C_7H_8$	316	0.9602124
C <sub>8</sub> H <sub>10</sub>	370	0.4610216
C <sub>8</sub> H <sub>10</sub>	375	0.8209473
C <sub>8</sub> H <sub>10</sub>	378	0.9408941
$C_{9}H_{12}$	435	5.233695
C <sub>9</sub> H <sub>12</sub>	435	1.383323
$C_{9}H_{12}$	430	0.5280889

#### **For Aromatics**

#### Appendix C Results of Investigation for Critical Pressure <u>Critical Pressures for Organic Components</u>

Component	P <sub>Cexp</sub> .	P <sub>Ccal.</sub> (bar)	Dev. %
	(bar)		
CH <sub>4</sub>	45.99	44.01	4.305
$C_2H_6$	48.72	47.9	1.683
C <sub>3</sub> H <sub>8</sub>	42.48	42.01	1.106
$C_{4}H_{10}$	37.96	37.1	2.265
$C_{4}H_{10}$	36.4	37.03	1.730
C <sub>5</sub> H <sub>12</sub>	33.7	33.056	1.910
C <sub>5</sub> H <sub>12</sub>	33.81	34.02	0.621
C <sub>5</sub> H <sub>12</sub>	31.96	31.6	1.126
$C_{6}H_{14}$	30.25	30.1	0.495
$C_{6}H_{14}$	30.1	29.89	0.697
$C_{6}H_{14}$	31.2	31.05	0.4807
$C_{6}H_{14}$	30.8	30.3	1.623
$C_{6}H_{14}$	31.3	30.99	0.9904
$C_{7}H_{16}$	27.4	27.012	1.416
C <sub>7</sub> H <sub>16</sub>	27.7	27.4	1.083
C <sub>7</sub> H <sub>16</sub>	27.4	27.06	1.240
$C_7H_{16}$	29.5	28.97	1.796
$C_7H_{16}$	29.1	28.95	0.515
$C_7H_{16}$	27.3	26.91	1.428
C <sub>7</sub> H <sub>16</sub>	28.1	29.99	6.725
$C_{7}H_{16}$	28.9	28.7	0.692
$C_{7}H_{16}$	29.5	29.2	1.016
C <sub>8</sub> H <sub>18</sub>	24.9	24.5	1.606
C <sub>8</sub> H <sub>18</sub>	28.7	28.6	0.3484
C <sub>8</sub> H <sub>18</sub>	24.8	24.7	0.4032
C <sub>8</sub> H <sub>18</sub>	25.5	25.035	1.823
C <sub>8</sub> H <sub>18</sub>	25.4	25.35	0.196
C <sub>9</sub> H <sub>20</sub>	23.5	23.43	0.297
C <sub>8</sub> H <sub>18</sub>	26.3	26.05	0.9505
C <sub>8</sub> H <sub>18</sub>	25.6	25.54	0.3243
C <sub>8</sub> H <sub>18</sub>	24.9	24.87	0.1204

## For Paraffine compounds

C <sub>8</sub> H <sub>18</sub>	26.9	26.92	0.0743
C <sub>8</sub> H <sub>18</sub>	72.3	72.12	0.2489
C <sub>8</sub> H <sub>18</sub>	28.2	28.045	0.5496
C <sub>8</sub> H <sub>18</sub>	27.3	27.4	0.366
C <sub>9</sub> H <sub>20</sub>	22.9	23.01	0.4803
C <sub>9</sub> H <sub>20</sub>	23.1	22.96	0.6060
C <sub>9</sub> H <sub>20</sub>	25.3	25.04	1.027
C <sub>9</sub> H <sub>20</sub>	24.9	24.81	0.3614
$C_{10}H_{22}$	21.1	20.94	0.758
$C_{10}H_{22}$	23.2	23.4	0.862
$C_{11}H_{24}$	19.8	20.03	1.16
$C_{12}H_{26}$	18.2	18.5	1.64
$C_{13}H_{28}$	16.8	16.7	0.595
$C_{14}H_{30}$	15.7	15.65	0.3184
C <sub>15</sub> H <sub>32</sub>	14.8	14.57	1.55
C <sub>16</sub> H <sub>34</sub>	14.0	14.3	2.14
C <sub>17</sub> H <sub>36</sub>	13.4	13.6	1.49

## for Olefins Compounds

Component	P <sub>Cexp.</sub>	P <sub>Ccal.</sub> (bar)	Dev. %
	(bar)		
C <sub>3</sub> H <sub>4</sub>	55.75	55.67	0.143
C <sub>4</sub> H <sub>8</sub>	40.2	40.3	0.248
C <sub>4</sub> H <sub>8</sub>	42.1	42.4	0.712
C <sub>4</sub> H <sub>8</sub>	41.0	40.99	0.024
C5H10	35.6	35.4	0.5617
C5H10	36.9	36.84	0.162
C <sub>5</sub> H <sub>10</sub>	38.6	38.45	0.388
C <sub>6</sub> H <sub>12</sub>	31.43	31.41	0.0636
C <sub>7</sub> H <sub>14</sub>	29.2	28.89	1.06
C <sub>7</sub> H <sub>16</sub>	29.1	28.78	1.09
C <sub>8</sub> H <sub>16</sub>	26.8	26.48	1.19
C <sub>9</sub> H <sub>18</sub>	23.3	23.01	1.24

## for Diolefins Compounds

Component	P <sub>Cexp.</sub>	<b>P</b> <sub>Ccal.</sub>	Dev. %
	(bar)	(bar)	
C <sub>3</sub> H <sub>4</sub>	52.5	51.987	0.977
C <sub>4</sub> H <sub>6</sub>	43.2	43.1	0.2314

## for Acetylenes Compounds

Component	P <sub>Cexp</sub> .	<b>P</b> <sub>Ccal.</sub>	Dev. %
	(bar)	(bar)	
$C_2H_2$	61.14	61.03	0.1799
C <sub>4</sub> H <sub>8</sub>	40.2	41.05	2.11

## for Aromatics Compounds

Component	P <sub>Cexp.</sub>	P <sub>Ccal</sub> .	Dev. %
	(bar)	(bar)	
C <sub>6</sub> H <sub>6</sub>	48.95	48.82	0.2655
C <sub>7</sub> H <sub>8</sub>	41.08	41.1	0.0486
C <sub>8</sub> H <sub>10</sub>	37.32	37.4	0.214
C <sub>8</sub> H <sub>10</sub>	35.41	35.1	1.94
C <sub>8</sub> H <sub>10</sub>	35.11	34.97	0.398
C <sub>9</sub> H <sub>12</sub>	34.54	34.35	0.55
C <sub>9</sub> H <sub>12</sub>	32.32	32.2	0.43
C <sub>9</sub> H <sub>12</sub>	31.27	31.4	0.415

Polar compound	P <sub>Cexp</sub> .	<b>P</b> <sub>Ccal</sub> .	Dev. %
	(bar)	(bar)	
H2O	221.2	220.1	0.497
NH <sub>3</sub>	113.5	114.45	0.837
HCL	83.1	82.9	0.24
CH <sub>3</sub> OH	80.9	81.6	0.865
С2Н5ОН	61.4	60.99	1.279
Ethylene Oxide	71.9	70.98	1.279
Acetone	47.0	46.89	0.234
I- Propanol	42.5	42.9	0.9411
N - Propanol	51.7	52.01	0.599
CH3CL	67.0	67.5	0.746
CH3F	56	56.9	1.607
<b>R</b> <sub>12</sub>	41.4	41.89	1.183
<b>R</b> <sub>11</sub>	44.1	45.03	2.108
<b>R</b> <sub>22</sub>	49.7	48.85	1.71
<b>R</b> <sub>32</sub>	58.3	57.78	0.891
<b>R</b> <sub>124</sub>	37.2	37.94	1.98

## for Polar Compounds

#### الخلاصة

التنبؤ بالقيم الدقيقة للخواص الحرجة لأي مركب نقي مهم جداً لأنها تدخل غالباً في حساب الخواص الفيزيائية المستعملة في عمليات التصميم الكيمياوية .القياس المختبري للخواص الحرجة لأي مركب هي عملية صعبة ومعقدة ومكلفة في نفس الوقت , لهذا السبب ولقياس الخواص الحرجة بصورة دقيقة فإن الاهتمام قد وجه لقياس هذه الخواص , وغالباً ما تقاس بطريقة المجموعات المساهمة , والتي تعتبر طريقة صعبة ومعقدة.

ولتفادي هذه المشكلة فإن الجهود قد بذلت من اجل تحسين أو تعديل المعدلات المتوفرة لحساب الخواص الحرجة للمركبات النقية من دون اللجوء إلى استعمال طريقة المجموعات المساهمة باستعمال طريقة سهلة نسبياً.

الطريقة المقترحة في هذا البحث, درجة الحرارة الحرجة, الحجم الحرج والضغط الحرج يمكن أن تحسب من بيانات درجة الغليان الطبيعية والوزن الجزيئي للمركب وذلك عن طريق التكرار حتى تتلاقى أو تقل نسبة الخطأ بالنسبة للضغط الحرج.

الطريقة المتبعة في هذا البحث يمكن أن تلخص بالخطوات التالية:

- 1. حساب درجة الحرارة الحرجة .
- 2. افتراض قيمة مناسبة للضغط الحرج وحساب الحجم الحرج .
  - 3. حساب عامل الانضغاطية الحرج .
  - 4. حساب الضغط الحرج باستعمال المعادلة التالية :

$$P_c = \frac{Z_c R T_c}{V_c}$$

اذا كمان الضغط الحرج النماتج من المعادلة أعلاه يختلف عن الضغط الحرج المفروض في الخطوة الثانية فإن الخطوات تكرر إلى أن تثبت قيمة الضغط الحرج وبنسبة خطأ تصل إلى 10-4 .

 استعمل برنامج إحصائي لإيجاد معادلة مناسبة وسهلة لقياس درجة الحرارة الحرجة للمركبات القطبية وغير القطبية باستعمال درجة الغليان الطبيعية والوزن الجزيئي للمركبات النقية ، والمعادلة هي كالتالي :

$$Tc = -11.5565 - 1.03586Mwt + 2.075167Tb - 0.000281Mwt^{2} - 0.00131Tb^{2} + 0.001827MwtTb$$

- معادلة رايدل لحساب الحجم الحرج استعملت بعد فرض قيمة مناسبة للضغط الحرج. النسبة المئوية لمعدل الخطأ لحساب الحجم الحرج بطريقة الأربع خطوات هي 1.7651% لـ 78 مركب نقي من مجموعات كيميائية مختلفة .
- 3. معادلة بيتزر لحساب عامل الانضغاطية الحرج قد استعملت للمركبات الغير قطبية ، أما بالنسبة للمركبات القطبية فقد استعملت معادلة وو و ستيل بعد تحسينها و هي كالتالي :

$$Z_c = 0.2901 - 0.879 * \omega - 0.033583 Y$$

طريقة لي - كيسلر استعملت لحساب عامل اللامركزية .

- 4. قيم نتائج الضغط الحرج المحسوب بطريقة الاربع خطوات هي قيم مرضية. النسبة المئوية لمعدل الخطأ لحساب الضغط الحرج لـ 114 مركب غير قطبي هي 1.98% و النسبة المئوية لمعدل الخطأ لحساب الضغط الحرج لـ 16 مركب قطبي هي 1.4671% .
- 5. النتائج أعلاه تشير بأن طريقة الأربع خطوات المقترحة لهذا البحث هي طريقة أكثر دقة من الطرق الاخرى المتوفرة لحساب الخواص الحرجة بضمنها طريقة المجموعات المساهمة ، هذا بالإضافة إلى حقيقة بأنها طريقة سهلة نسبيا إذا ما قورنت بطريقة المجموعات المساهمة.

## شکر وتقدیر

أتقدم بشكري وإمتناني الكبيرين إلى أساتذتي الأعزاء في قسم الهندسة الكيمياوية وخصوصا الأب الأول أستاذي الفاضل دمحمود عمر عبد الله الذي أسدى لي النصيحة ومد يد العون في إعداد هذا البحث والي قسم الهندسة الكيمياوية لما قدمه لي من تسهيلات ومصادر علمية متمنية دوام الموفقية والنجاح للمسيرة العلمية في وطننا العزيز.

## التحقق من الارتباطات للتنبؤ بالخواص الحرجة للمركبات النقية

# رسالة مقدمة إلى كلية الهندسة في جامعة النهرين وهي جزء من متطلبات نيل درجة ماجستير علوم في الهندسة الكيمياوية



ربيع الأول تشرين الثاني ٢٠٠٧